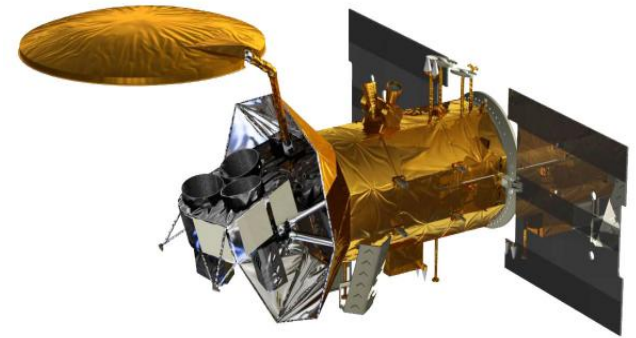


Microwave Radiometer (MWR) Counts to Tb (Brightness Temperature) Algorithm Development (Version 6.0) and On-Orbit Validation



Zoubair Ghazi

CFRSL – Central Florida Remote Sensing Lab

Dissertation Defense
October 24, 2014

Dissertation Objective

- To develop an improved counts to brightness temp (Tb) algorithm for the CONAE Microwave Radiometer on the Aquarius/SAC-D satellite
- Validation of Tb measurements using inter-satellite radiometric comparisons (X-CAL)
- Produce an Algorithm Theoretical Basis Document (ATBD) and deliver prototype MatLab code to CONAE

MWR Counts-to-Tb Algorithm History

- Post-launch CFRSL & CONAE evaluated MWR counts-to-Tb algorithm V5.0
 - Used 6 mo of MWR on-orbit collocation with WindSat
 - Ocean Tb' s exhibited small and acceptable Tb biases
 - Land Tb' s exhibited anomalous behavior
 - Land/water Tb transitions were “Smearred”
 - Step function changes of noise diode deflections
- Based upon on-orbit evaluation, it was concluded that:
 - V5.0 was unacceptable for producing MWR science data
 - An improved counts-to-Tb algorithm must be developed to address the anomalous Tb effects
- Further, CONAE developed a revised Counts-to-Tb algo V5.0S that included a smear correction
- **This was the starting point for my dissertation research**

My Research Approach

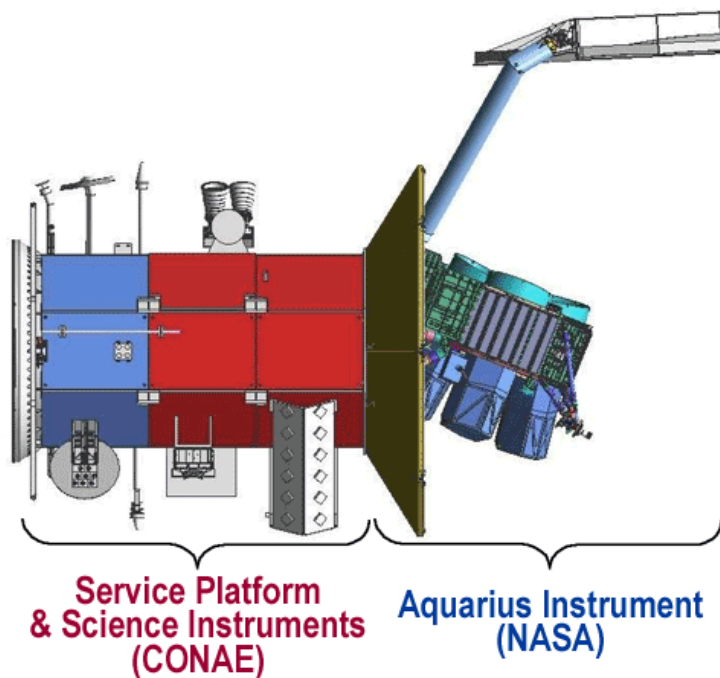
1. Evaluated the MWR counts-to-Tb algorithm V5.0S
 - On-orbit X-CAL with WindSat indicated that
 - Smear effects at land/water boundaries were removed
 - However, anomalous effect of noise diode deflections remained
 - Determined that MWR system gain varied with scene Tb
2. Developed a forward model for MWR system Counts-to-Tb
 - Empirically derived coefficients to match on-orbit observations, including deep space calibrations
 - Characterized model coefficients versus scene Tb
3. Developed a gain non-linearity correction
4. Implemented a new inverse model Counts-to-Tb algorithm V6.0
5. Validated algorithm using X-CAL with WindSat

Introduction to Aquarius/SAC-D and MWR

Aquarius/SAC-D Mission

Aquarius (AQ) is a mission of “Original Exploration” First NASA mission to measure Sea Surface Salinity (SSS) from space

SAC-D was launched on **June 10th , 2011** from Vandenberg Air Force Base, California.

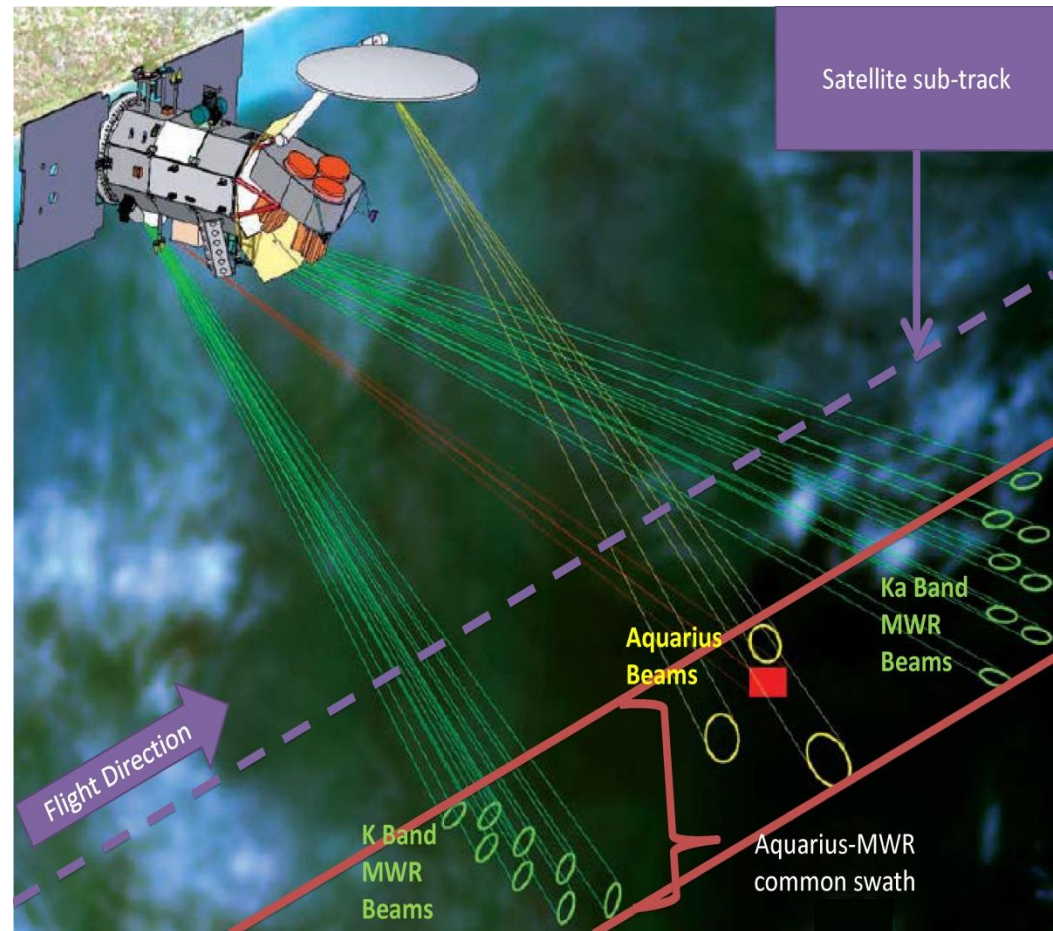


- Aquarius instrument - NASA
- **MWR - CONAE**
(Argentinian Space Agency)
- MWR provides auxiliary environment measurements: **water vapor, ocean surface wind speed, and oceanic rain rate**

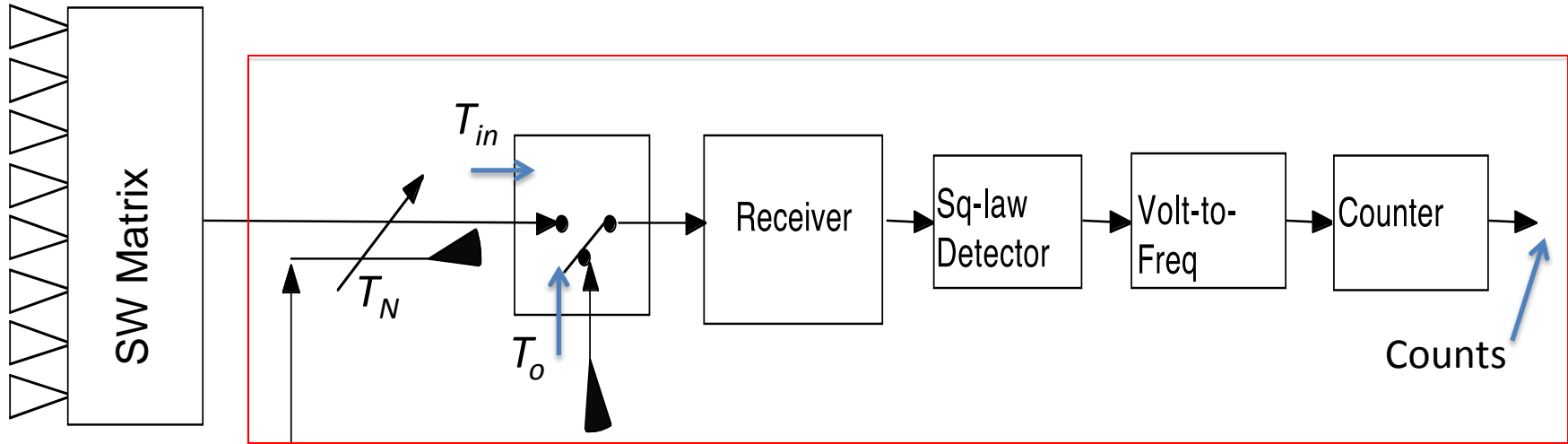
Microwave Radiometer - MWR

MWR supports AQ science by measuring simultaneous & collocated ocean brightness temperatures (T_b)

- 3 channel push-broom Dicke radiometer:
 - 36.5 GHz H- & V-Pol (forward-look)
 - 23.8 GHz H-Pol (aft-look)
- Earth Incidence angle
 - 52° for odd beams
 - 58° for even beams
- Matches the AQ swath width of 380 km



MWR Single Channel Block Diagram



Three Dicke radiometer states:

$$C_a = (T_{in} + T_{recv}) * G_{recv} + V_{off-set} \quad (1)$$

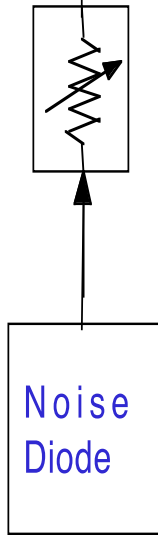
$$C_N = (T_{in} + T_N + T_{recv}) * G_{recv} + V_{off-set} \quad (2)$$

$$C_o = (T_o + T_{recv}) * G_{recv} + V_{off-set} \quad (3)$$

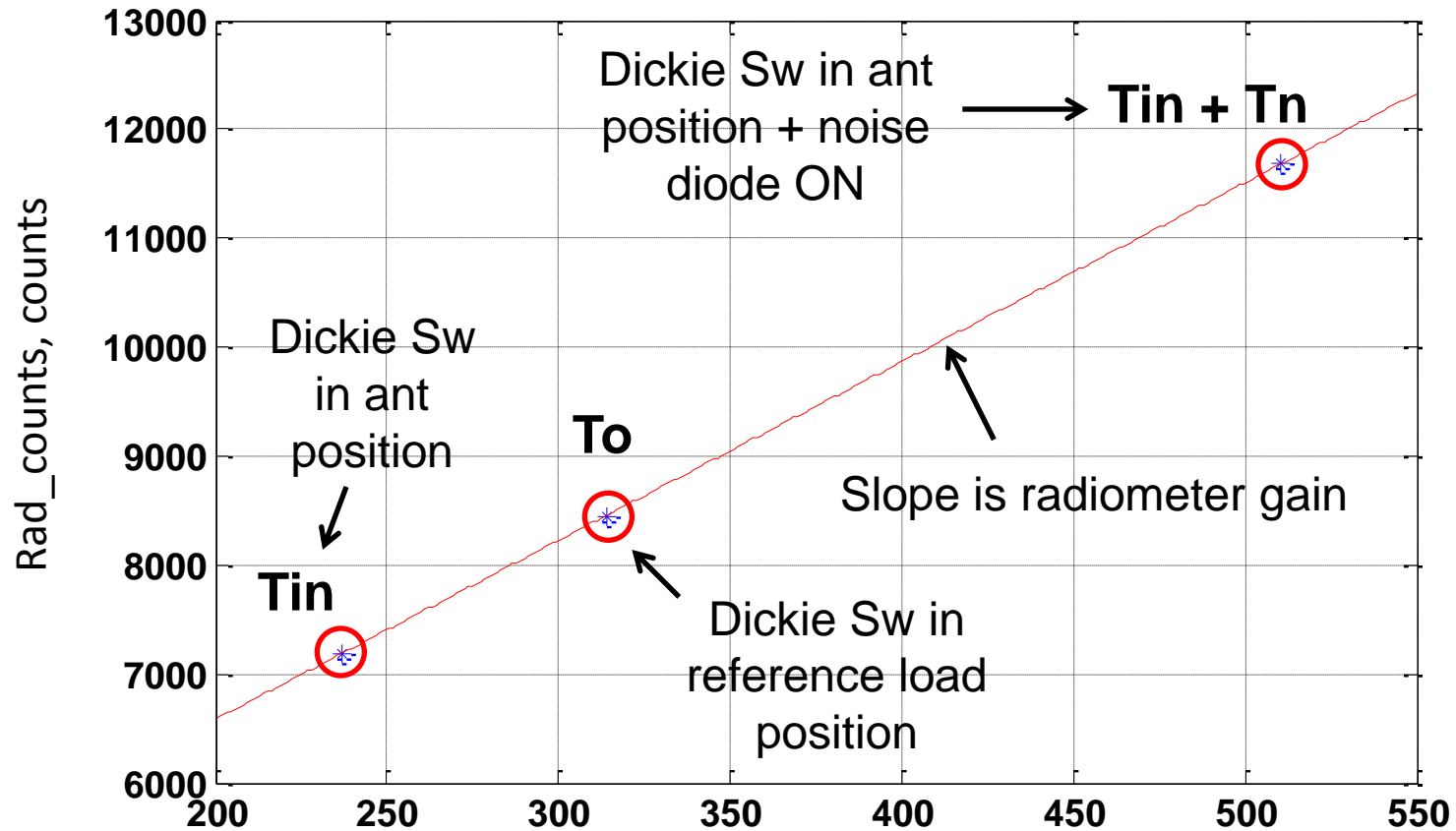
Subtracting (1) from (2) yields the radiometer gain, which varies in time

$$G_{recv} = \frac{C_N - C_a}{T_N} \quad T_{in} = \frac{C_a - C_o}{C_N - C_a} * T_N + T_o$$

On/off



Example: Dicke Radiometer Transfer Function (for constant gain)



Radiometer Input to the antenna port of Dicke switch, T_{in} , Kelvin

On-Orbit MWR Transfer Function

- Our objective was to determine the MWR transfer function based upon on-orbit measurements
- However, under typical on-orbit condition, the radiometer system gain will vary cyclically (once/orbit) due to the receiver physical temperature changes
- Therefore, a procedure was developed to synthesize **rad_counts @ constant system gain** from MWR measurements

Counts Normalization Procedure to Remove Variable Receiver Gain

- Time variable gain was removed and all counts were normalized using the following equation:

$$Co_{norm_i} = Co_i * \frac{\langle Gain \rangle}{Gain_i}$$

$$Co_i = (To_i + \langle Trec \rangle) * Gain_i$$

To_i is the instantaneous reference load physical temperature, Kelvin

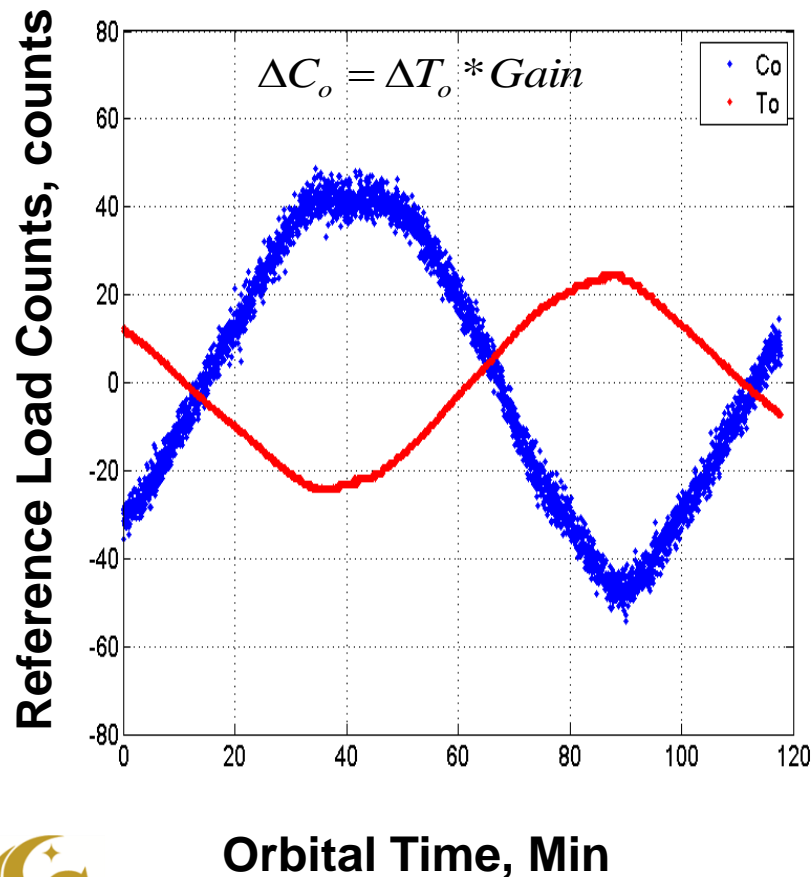
$\langle Trec \rangle$ is the orbit average receiver noise temperature

$Gain_i$ is the instantaneous system gain

$\langle Gain \rangle$ is the orbit average gain

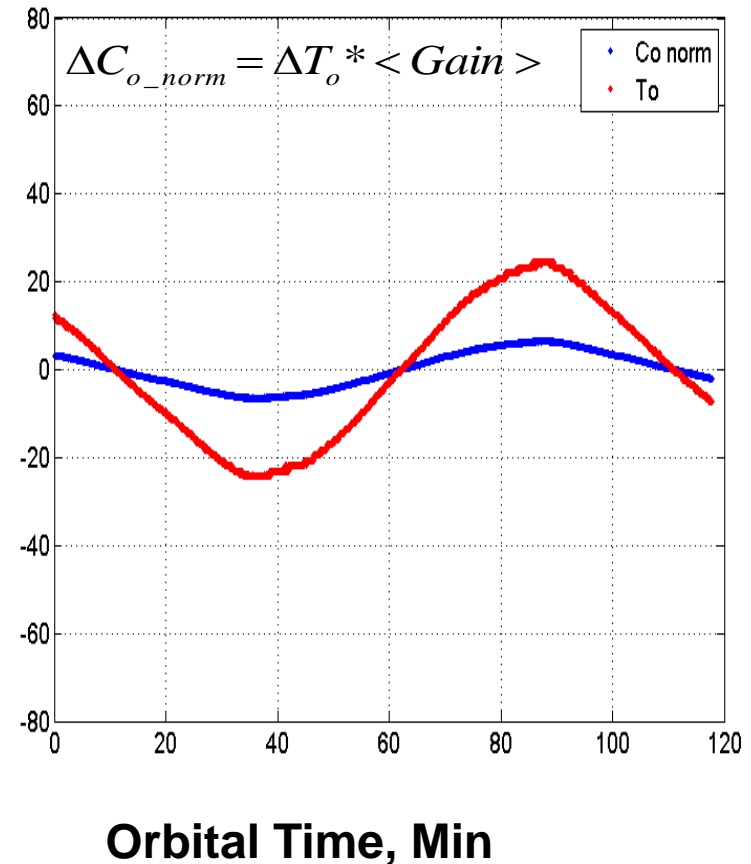
Example Gain Normalization

Before Normalization



After Normalization

Reference Load Temperature, K/100

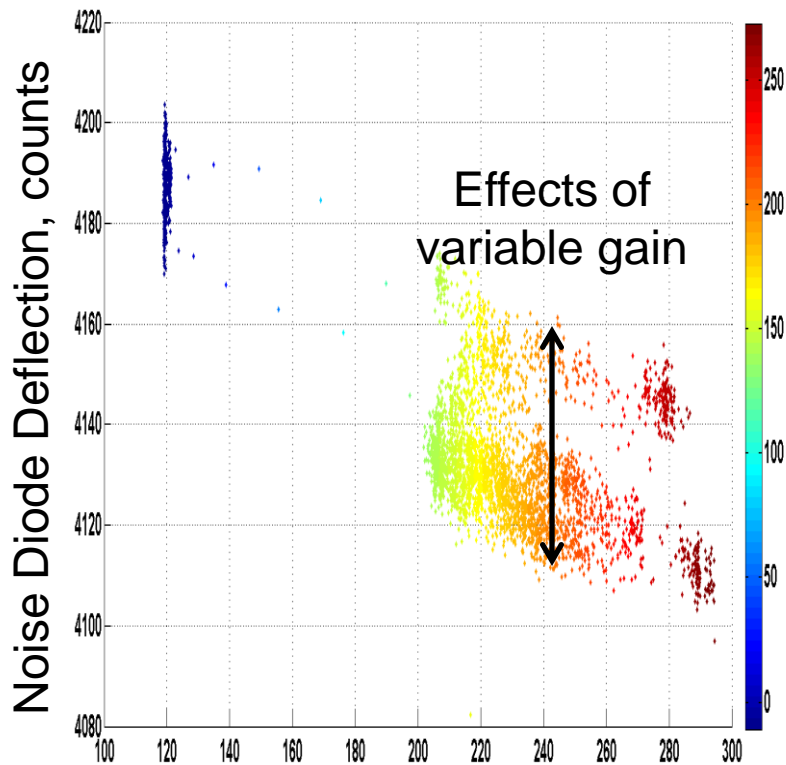


Measurement of MWR System Gain (Noise Diode Deflection)

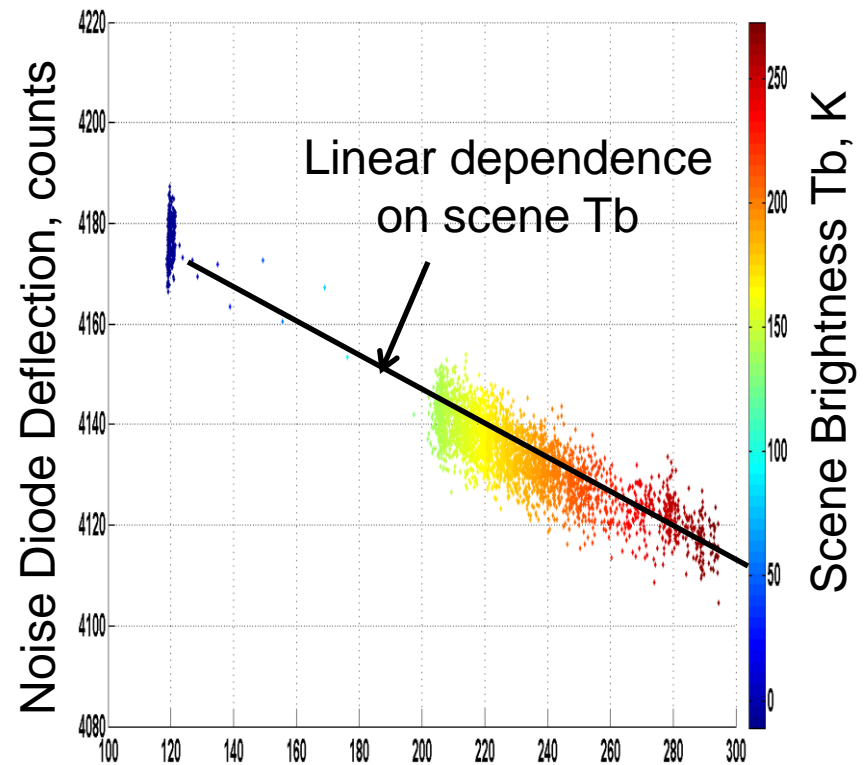
37GHz V-pol, Comparison of V5.0S

Noise Diode Deflection = $(C_n - C_a)$

Before Count (gain)
Normalization



After Count (gain)
Normalization



Radiometer Input to the antenna port of Dicke switch, T_{in} (Kelvin)

MWR Signal Processing with non-linear system gain compression

$$C_a = (T_{ant} + T_{recv}) * G_{recv-1} + V_{off-set} \quad (1)$$

$$C_N = (T_{ant} + T_N + T_{recv}) * G_{recv-3} + V_{off-set} \quad (2)$$

$$C_{ref} = (T_{ref} + T_{recv}) * G_{recv-2} + V_{off-set} \quad (3)$$

$$(T_{ant} + T_N + T_{recv}) > (T_{ref} + T_{recv}) > (T_{ant} + T_{recv})$$

therefore,

$$G_{recv-3} < G_{recv-2} < G_{recv-1}$$

MWR Slightly Non-linear Gain Model

$$G_{recv_1} = G_o g(T_{ref}) h_1(T_{in})$$

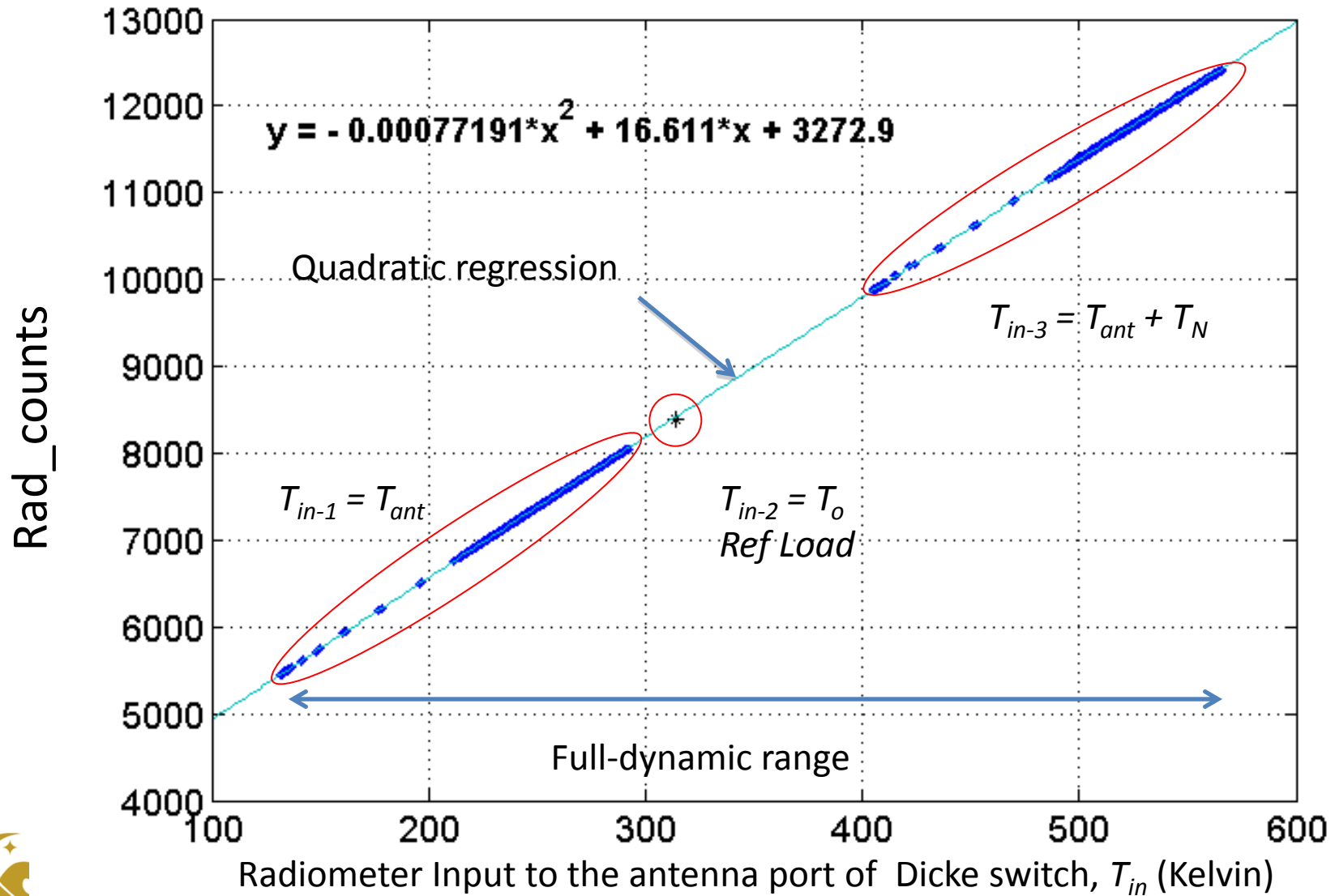
$$G_{recv_2} = G_o g(T_{ref}) h_2(T_{in})$$

$$G_{recv_3} = G_o g(T_{ref}) h_3(T_{in})$$

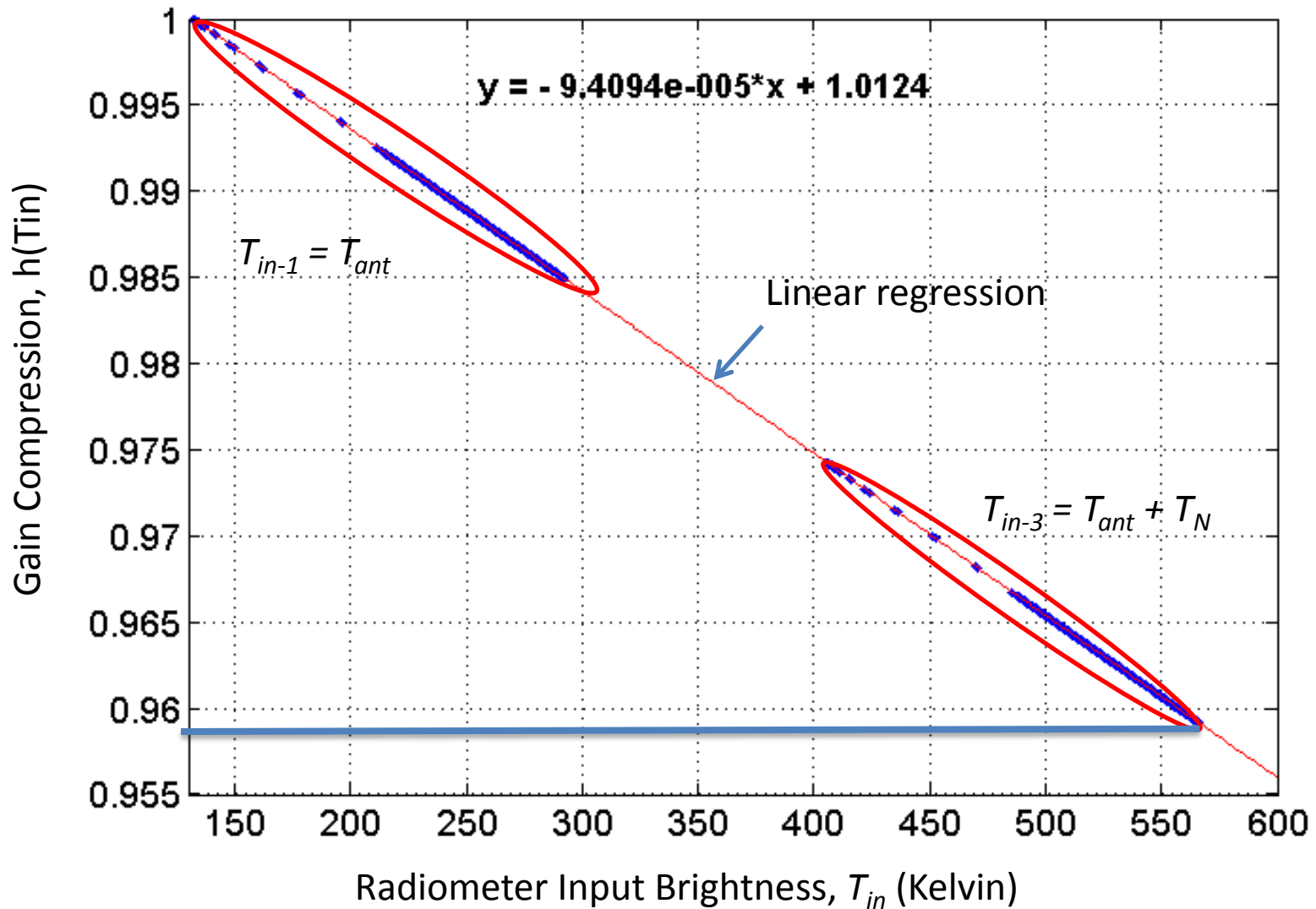
- G_o is the mean “long term gain”
- $g(T_{ref})$ is the orbital gain change due to phy temp (T_{ref})
- $h(T_{in})$ is the gain compression due to variable scene brightness temp (and injected noise diode)

These parameters are estimated during a single orbit where a deep-space calibration is performed

MWR Radiometer Transfer Function for One Orbit



Gain Compression Function, $h(T_{in})$

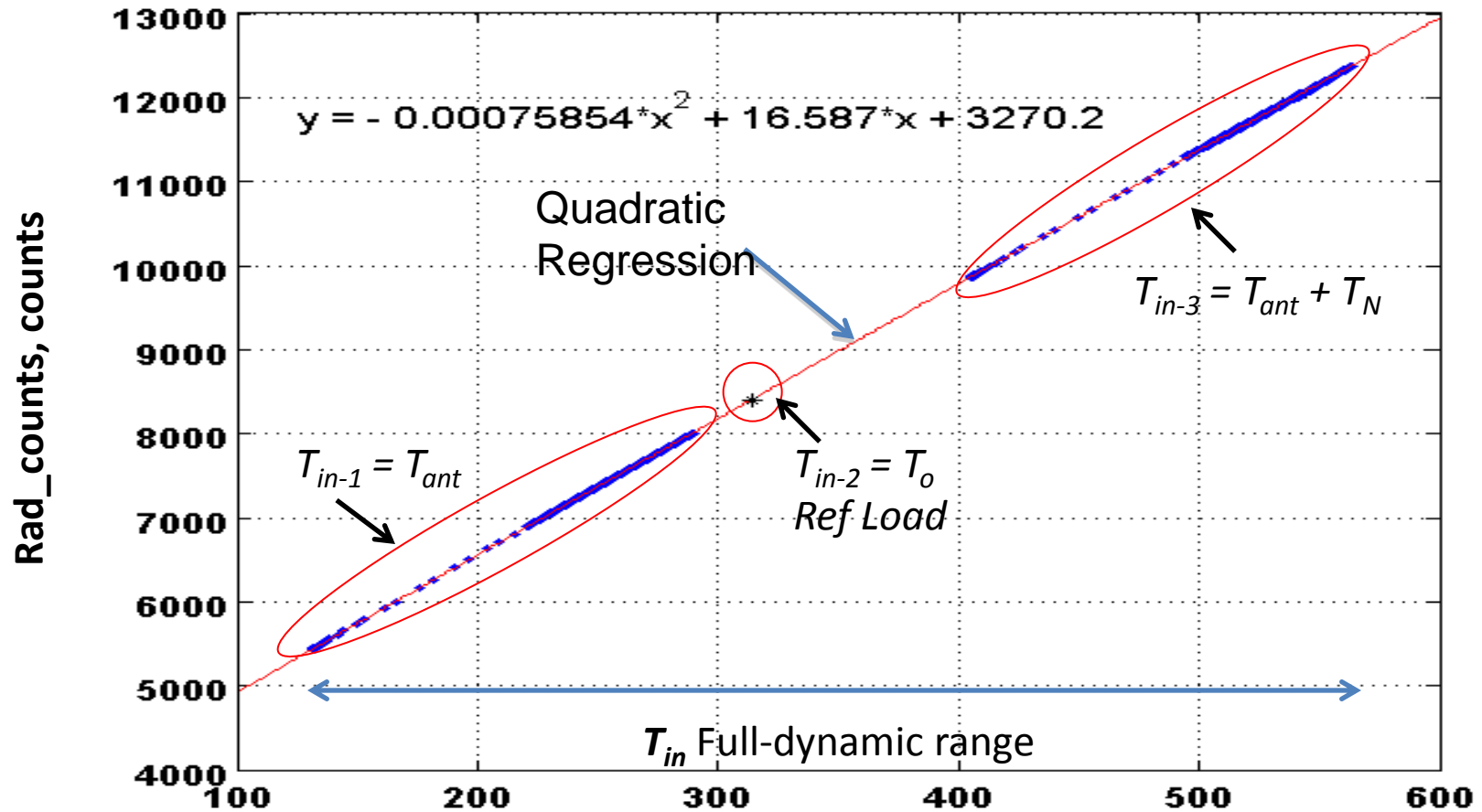


Non-Linearity Correction, Radiometer Transfer Function (V5.0S)

- Seven Deep Space Calibration (DSC) orbits that included, space, ocean, and land observations were used to cover wide range of scene T_b 's
- After counts (gain) normalization, the radiometer transfer function was established
 - $Rad_counts = f(T_{in})$
- Quadratic regression for 37V channel yielded the following

$$Rad_counts = -7.5 \times 10^{-4} (T_{in})^2 + 16.58 (T_{in}) + 3270$$

MWR Radiometer Transfer Function for 37V V5.0S (constant gain) for One Orbit



Radiometer Input to the antenna port of Dicke switch, T_{in} (Kelvin)

V6.0 Radiometer Transfer Function Linearization

- Averaging 2nd order regression coeff's from 7 DSC orbits, the instantaneous counts linearization equation is:

- For 37 V
$$C_{x_linear} = C_x - (-7.4677e - 004) * T_{in}^2$$

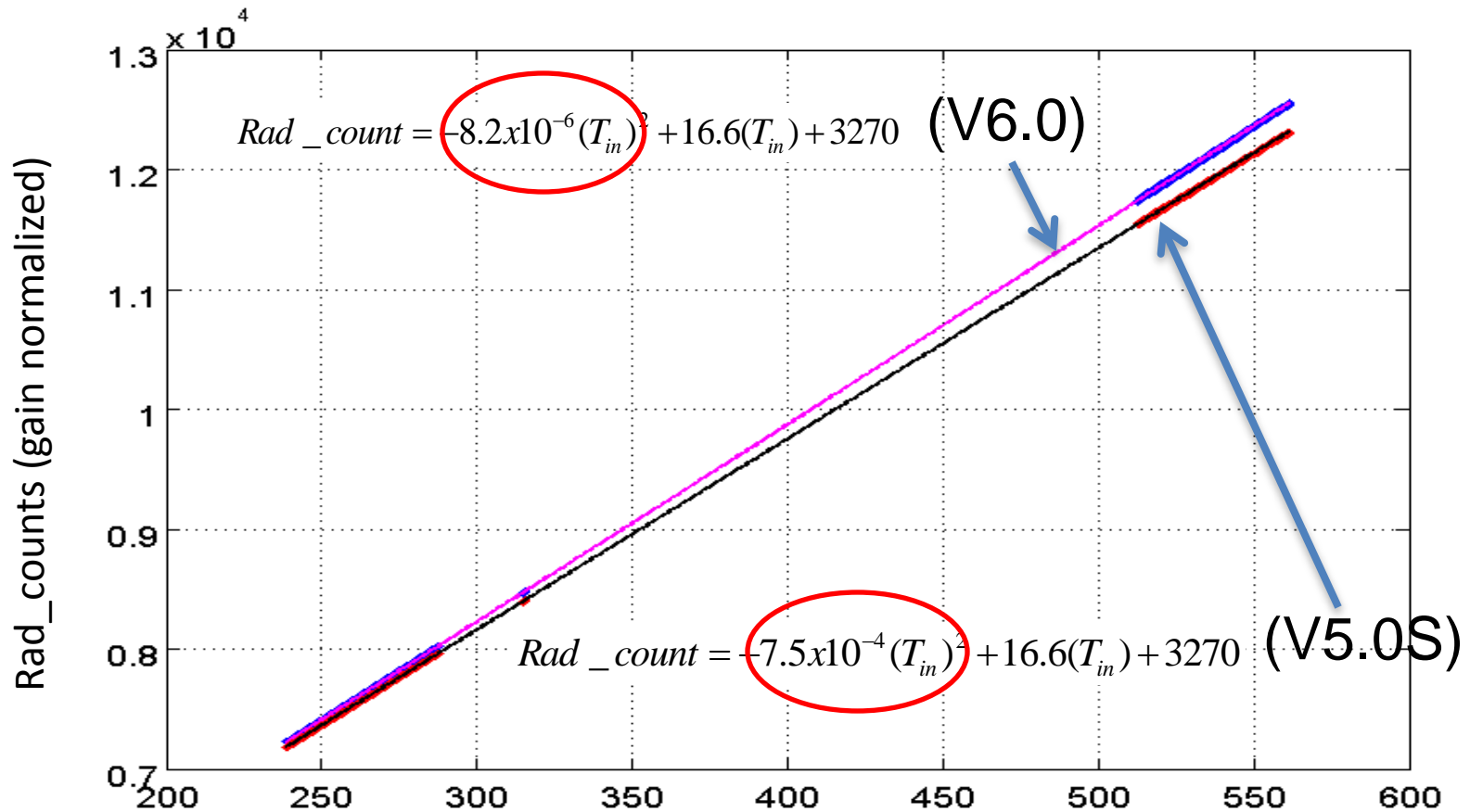
- For 37 H
$$C_{x_linear} = C_x - (-6.9064e - 004) * T_{in}^2$$

- For 23 H
$$C_{x_linear} = C_x - (-2.1708e - 004) * T_{in}^2$$

Where $x = ant, N, \text{ and } ref$

T_{in} is the input Tb to the Dicke switch, which is estimated using non-linear counts

Radiometer Transfer Function 37V (constant gain) V5.0S and V6.0



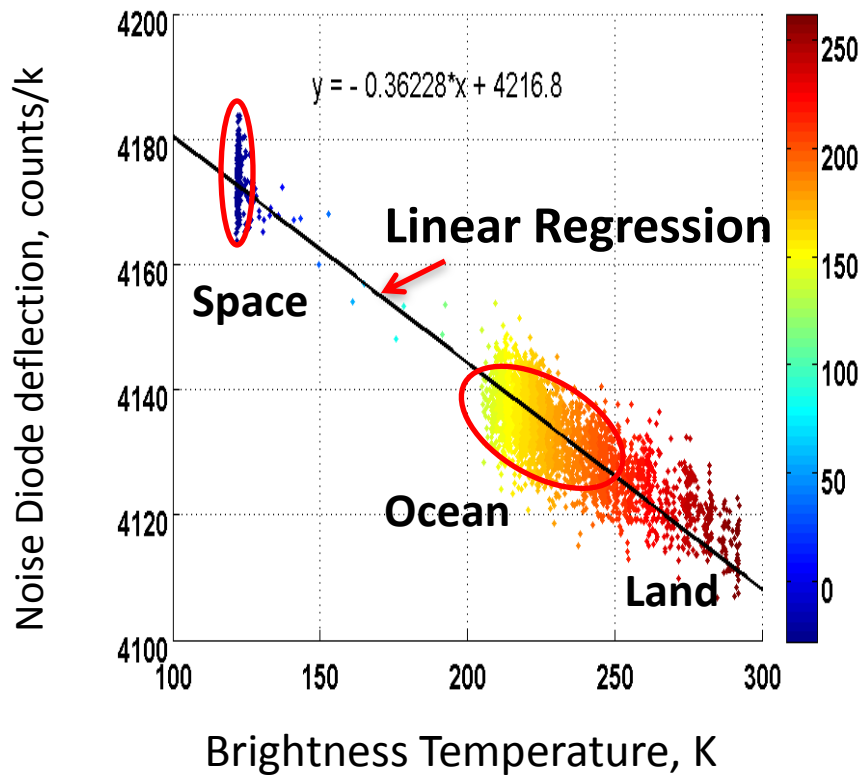
Radiometer Input to the antenna port of Dicke switch, T_{in} (Kelvin)

37GHz V-pol , One Orbit Noise Diode

Deflection = $(C_n - C_a)$

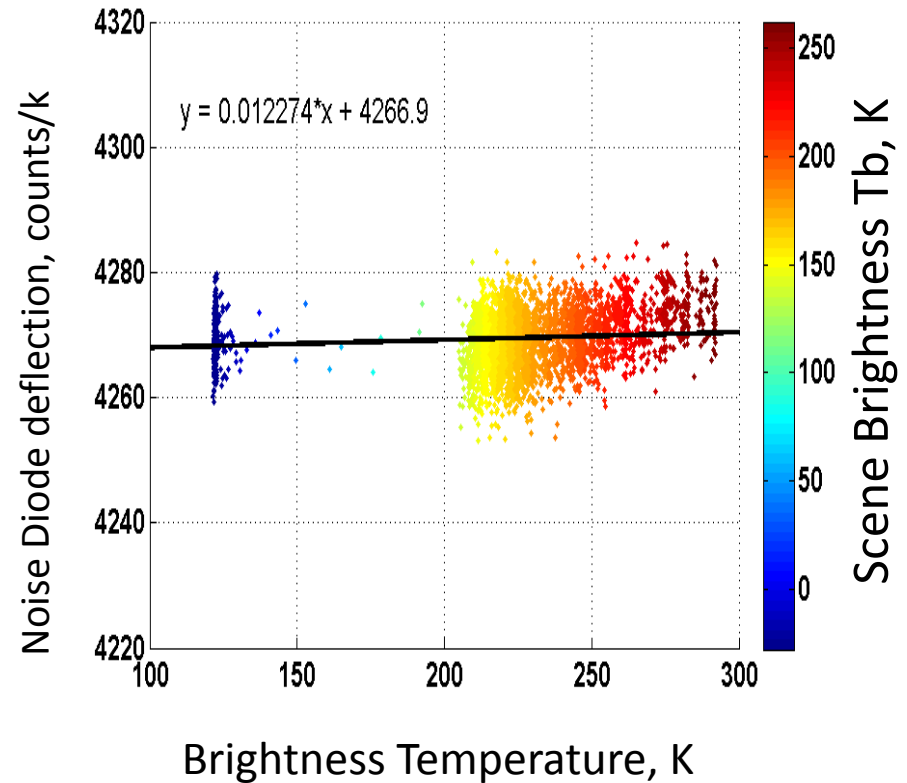
V5.0S

(Without counts linearization)



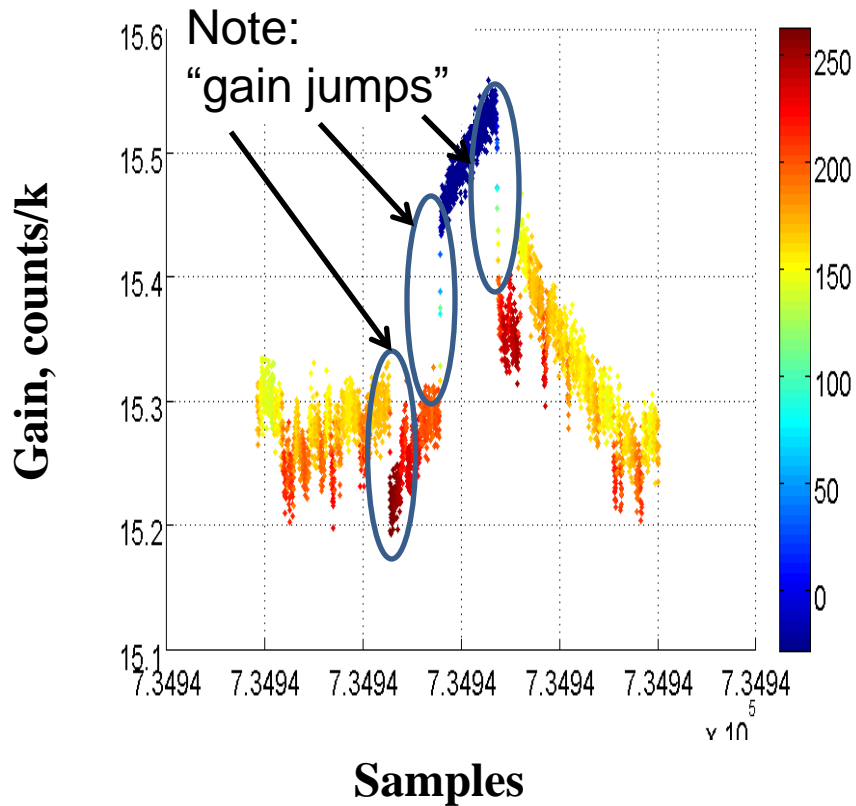
V6.0

(With counts linearization)

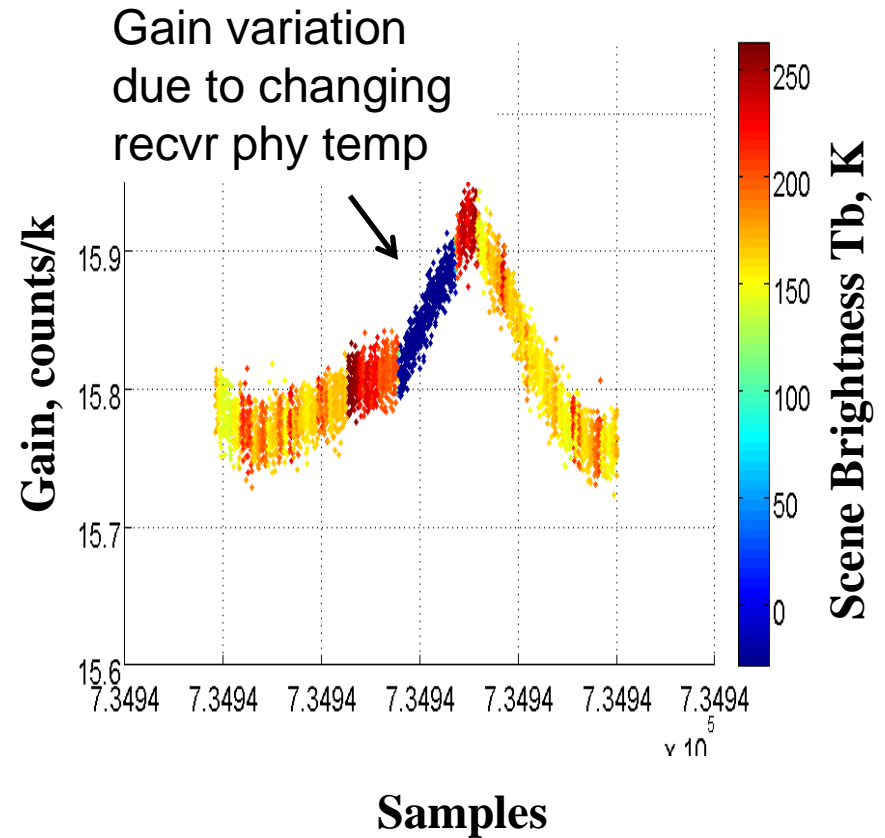


37GHz V-pol , One Orbit Radiometer Gain

V5.0S



V6.0

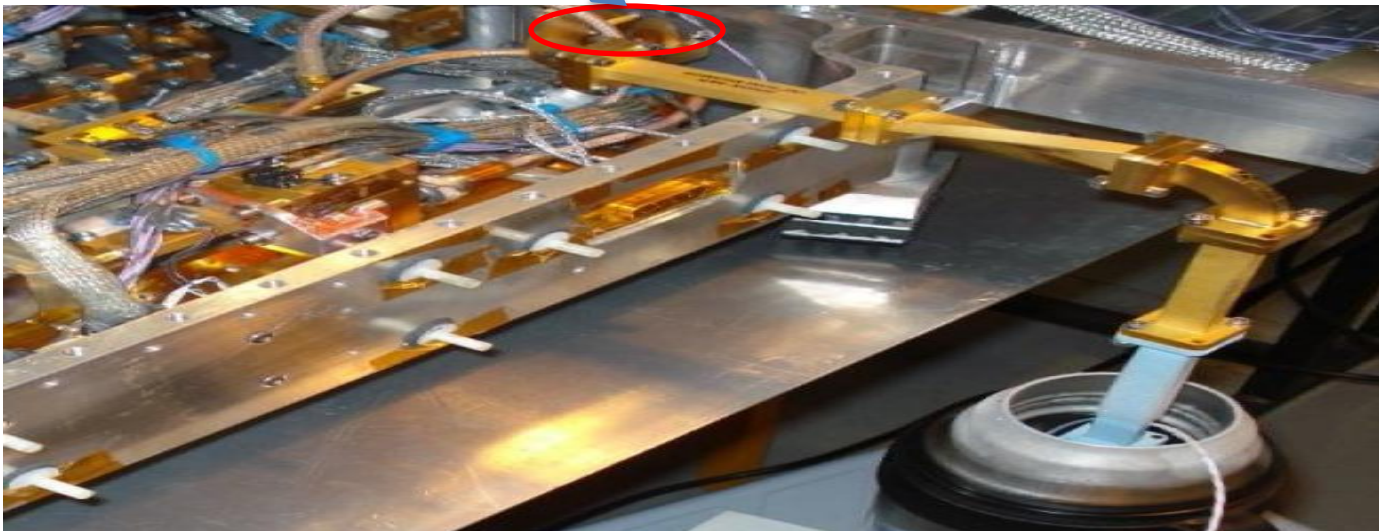
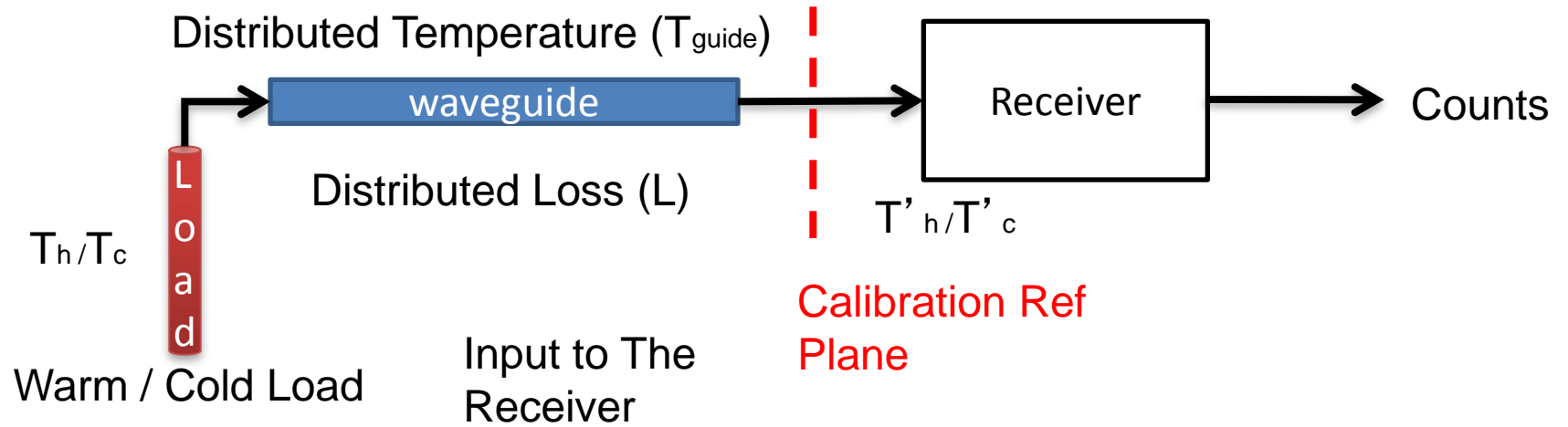


Reanalysis of Prelaunch TV Calibration Test using Linearized Counts

Pre-launch Data Analysis Objectives for V6.0 (using Linearized Counts)

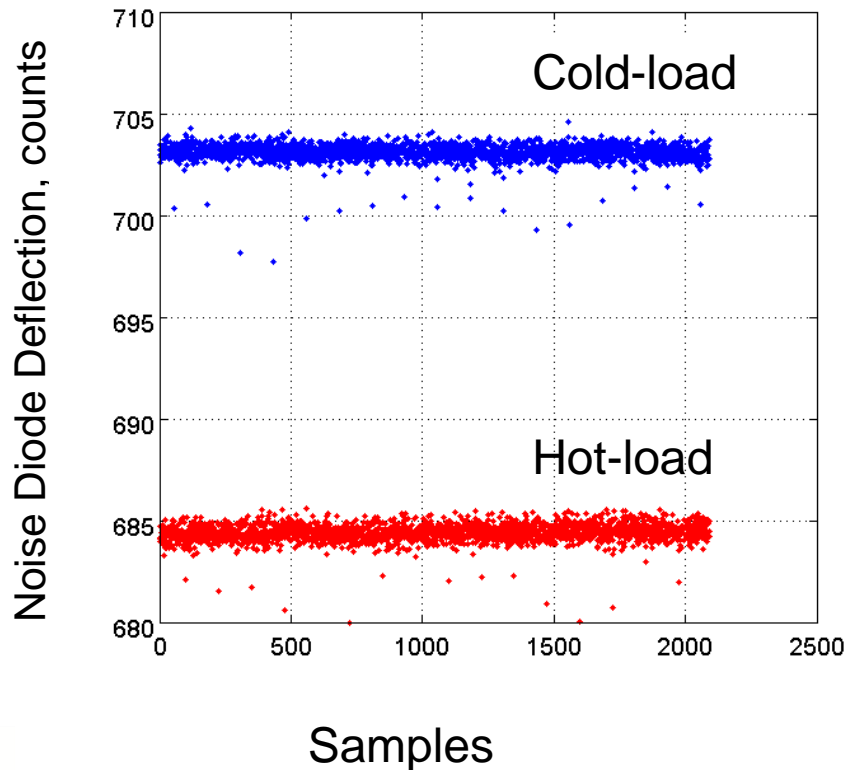
- Characterization of injected noise diode temperature (T_N) over physical temperature
- Retrieve **antenna switch matrix** loss coefficients
 - Empirical method (regression model) was applied
 - Assumption: All transmission and reflection coeff's are constant and are NOT expected to change during MWR's mission life time

Pre-TV test Block Diagram

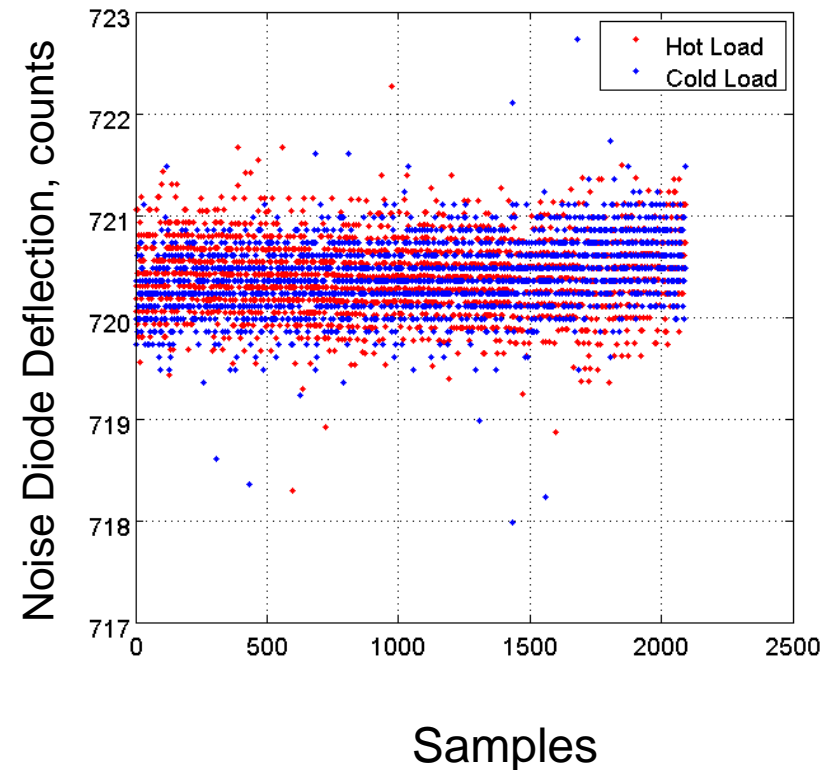


Pre-TV test, Comparison of the Noise Diode Deflection V2.0 and V6.0

SAYAK BISWAS
V2.0



V6.0



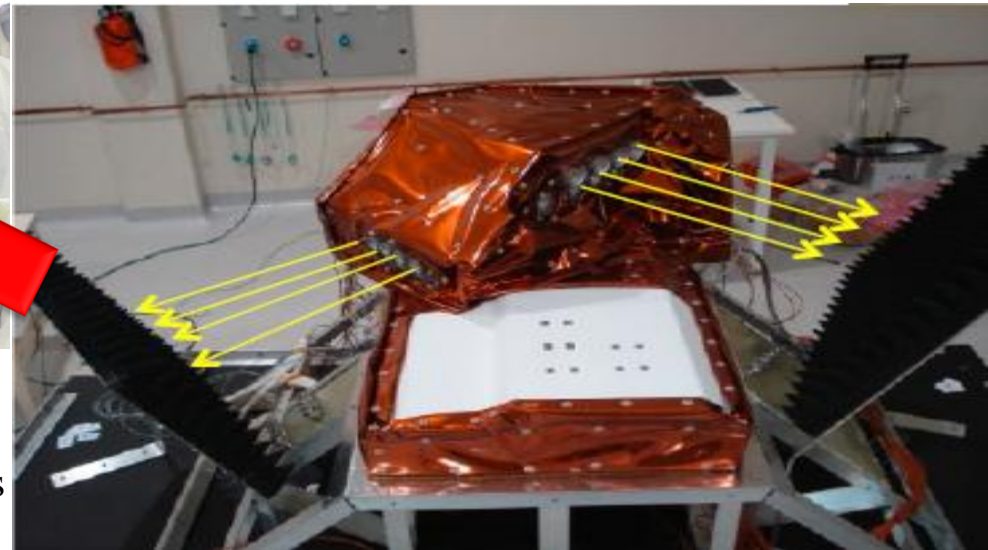
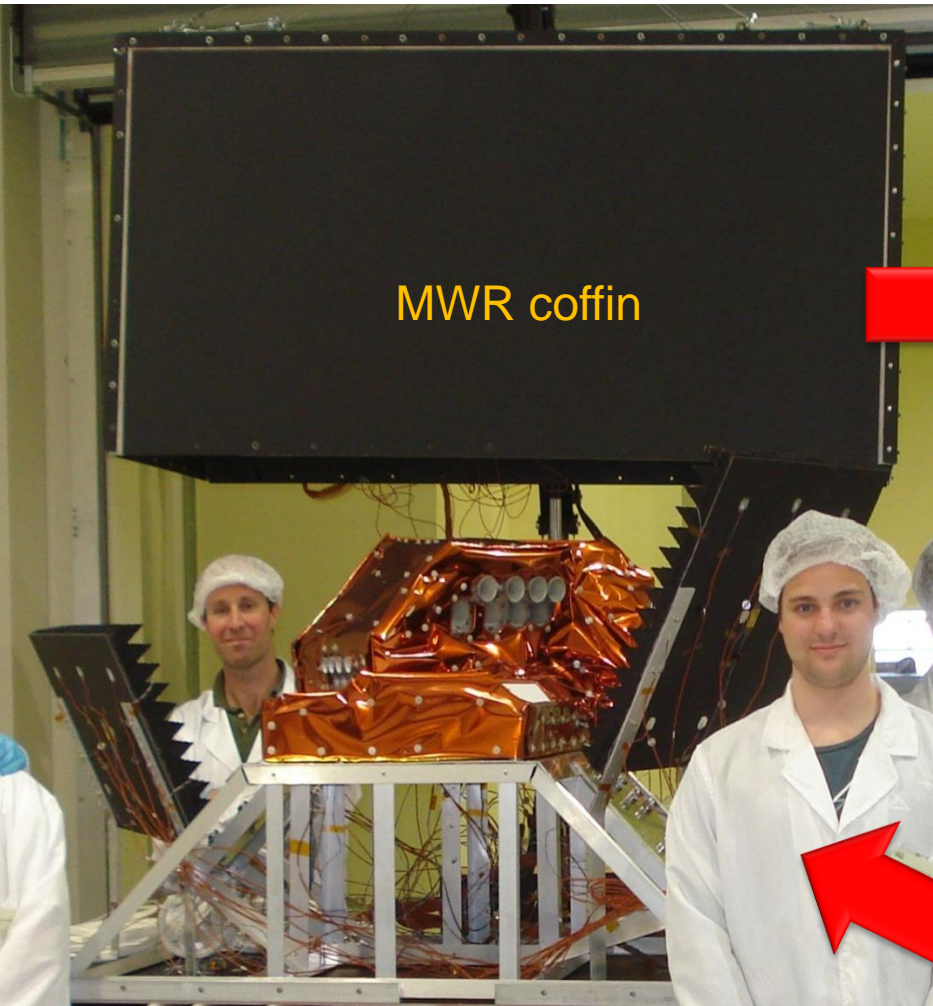
Inversion Model for Apparent Brightness Temperature (T_{ap})

$$T_{ap} = [T_{in} - (b_2 * T_0 + b_3 * T_1 + b_4 * T_2 + b_5 * T_3 + b_6 * T_4)] / b_1$$

– where

- T_{ap} is the scene brightness temp at horn aperture
- T_{in} is the input brightness temperature to antenna port of Dicke switch
- T_0, T_1, T_2, T_3 , & T_4 are MWR physical temps
- b_1, b_2, b_3, b_4, b_5 and b_6 are **antenna switch matrix** loss coefficients derived using the regression model

Thermal Vacuum Test



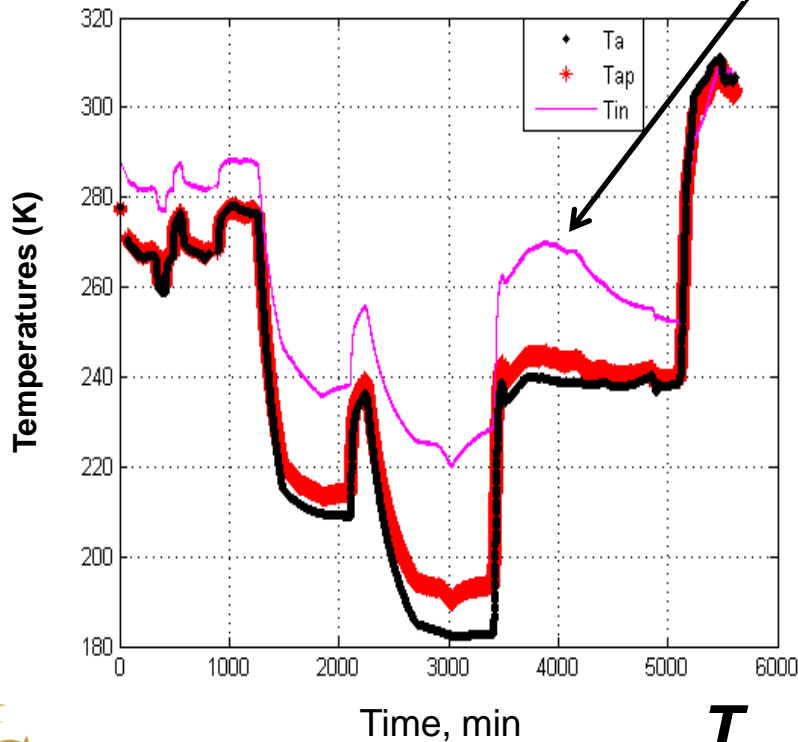
- The thermal vacuum (TV) test for MWR was performed in September 2009. (09/06 – 09/09)
- Performance of the SW matrix losses coefficients

37GHz V-pol, Computed Apparent Temp based on: SWM losses and reflection (V2.0), regression model (V6.0)

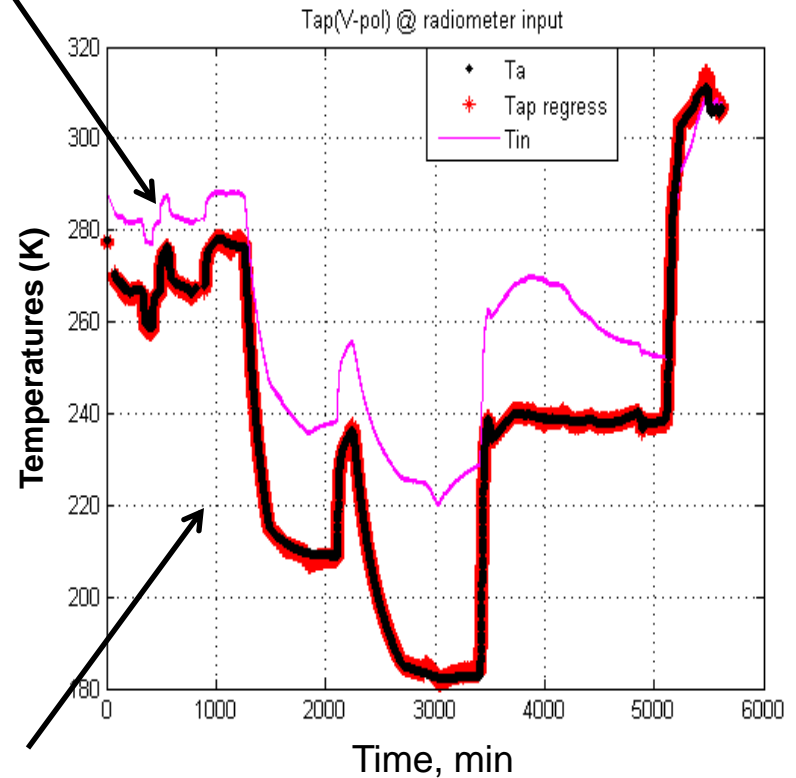
$$T_{ap} = [T_{in} - (b_2 * T_o + b_3 * T_1 + b_4 * T_2 + b_5 * T_3 + b_6 * T_4)] / b_1$$

T_{in} calc from rad_counts

SWM Model, RMS=3.94

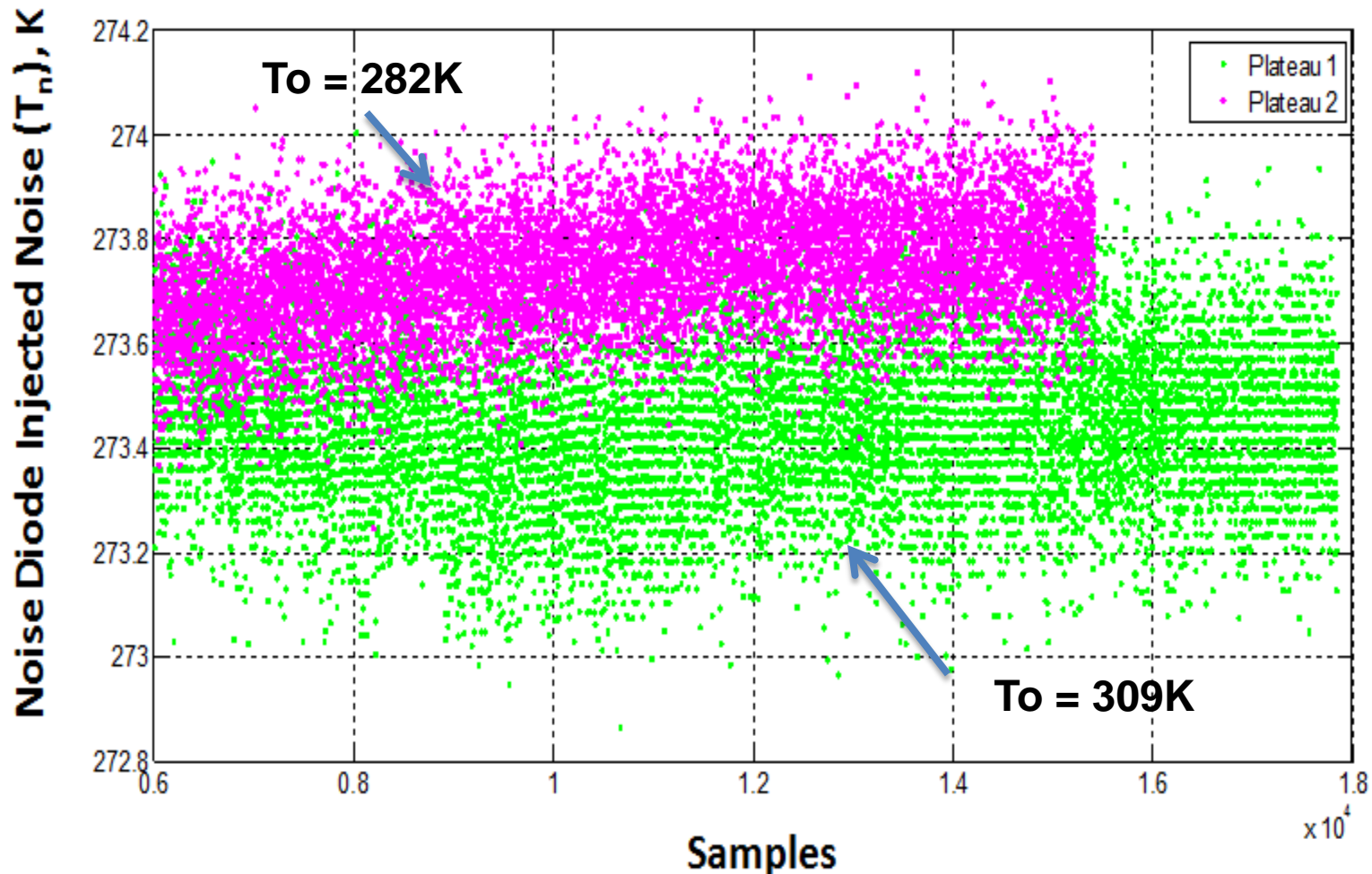


Regression Model, RMS= 0.75



T_{ap} & T_{ap_regress}

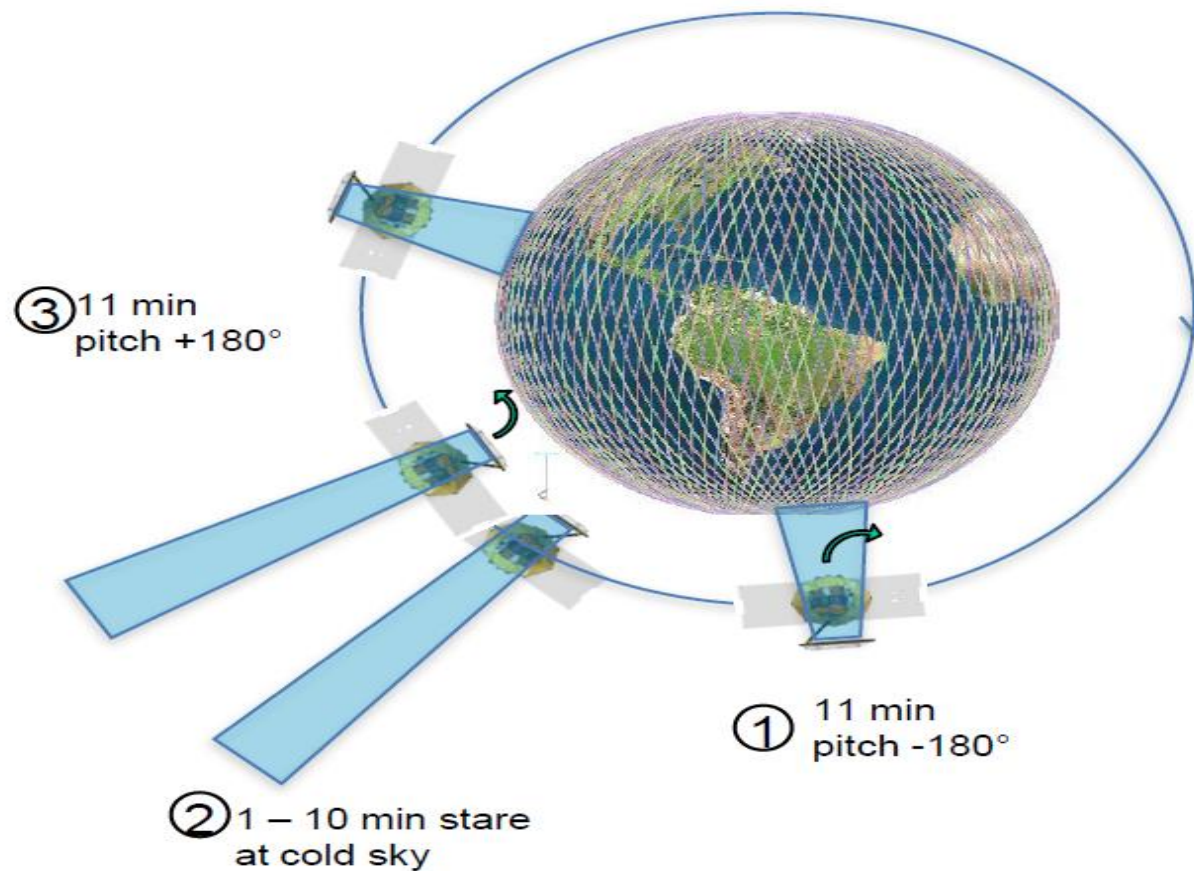
Calculated Noise Diode Injection Noise (T_N) for Two Different TV Plateaus



- Post-Launch MWR Calibration Analysis
 - Deep Space Calibration
 - WindSat X-CAL

Deep Space Calibration cause MWR Antenna Beams to view Homogeneous Scene = 2.7 K

View from Night side towards Sun



Post Launch X-CAL Analysis

- APC and residual bias correction were applied by inter-satellite XCAL
 - MWR = target & WindSat = reference
- MWR and WindSat have different incident angles, therefore, Tbs were adjusted using theoretical radiative transfer model values for both satellites (MWR_{sim} and WS_{sim})

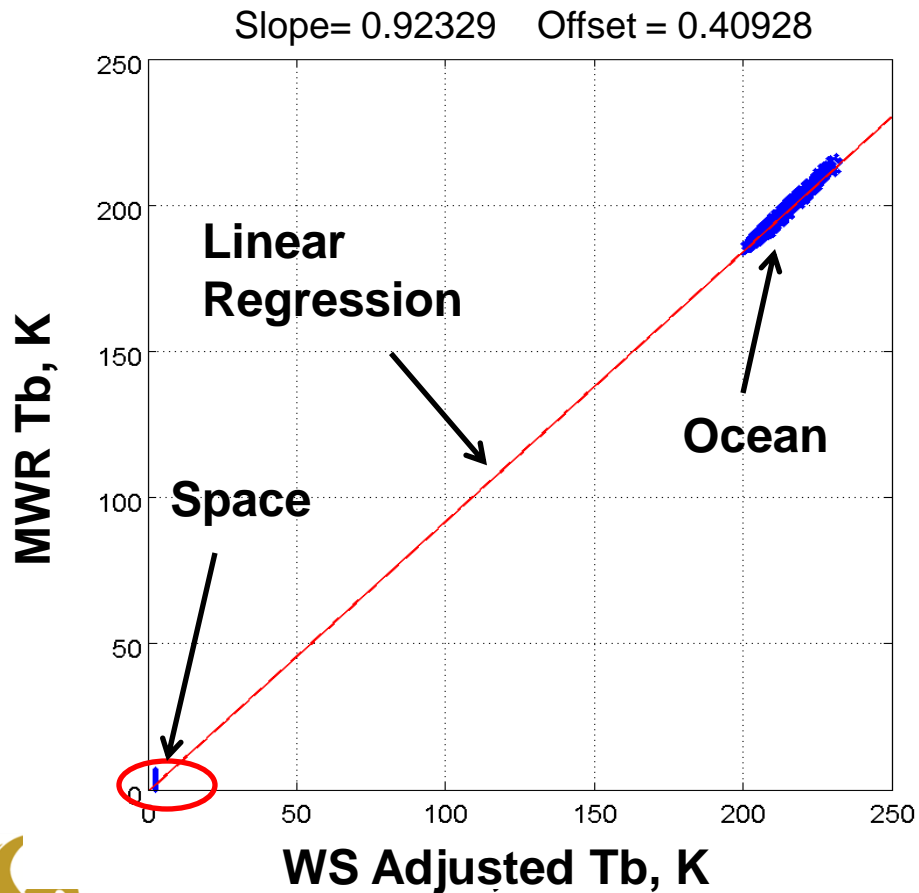
$$WS_{adj} = WS_{obs} + (MWR_{sim} - WS_{sim})$$

- Double Difference Technique

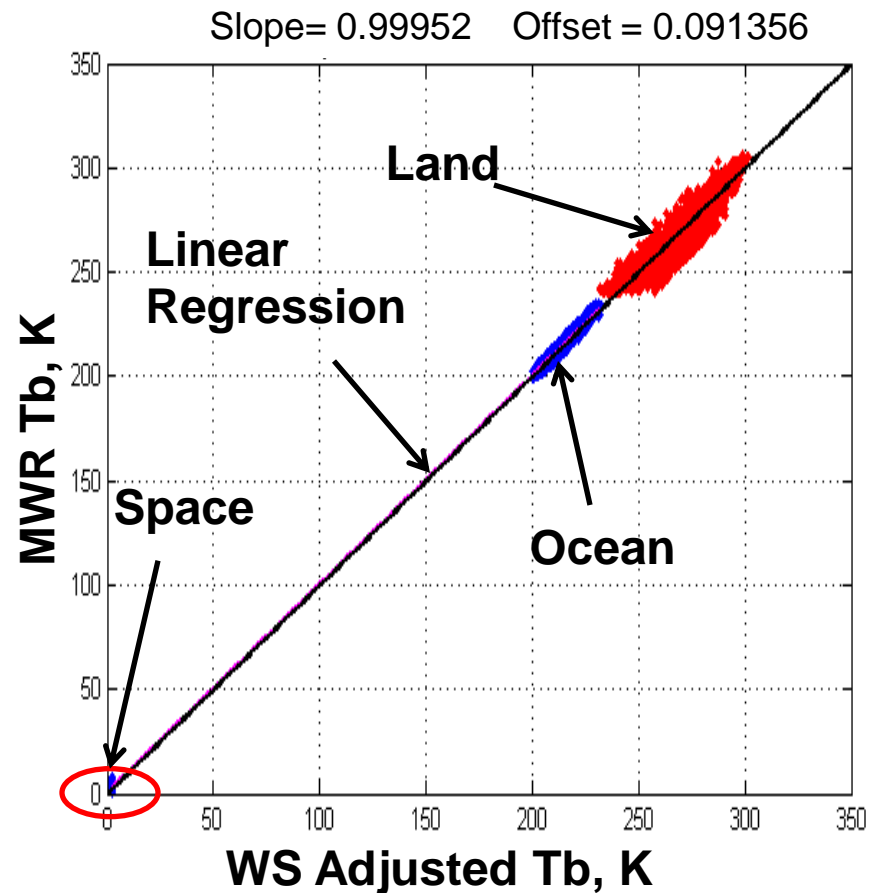
$$DD = MWR_{obs} - WS_{adj}$$

37GHz V-pol Beam-1: X-CAL of MWR V6.0 and WindSat

Before Correction

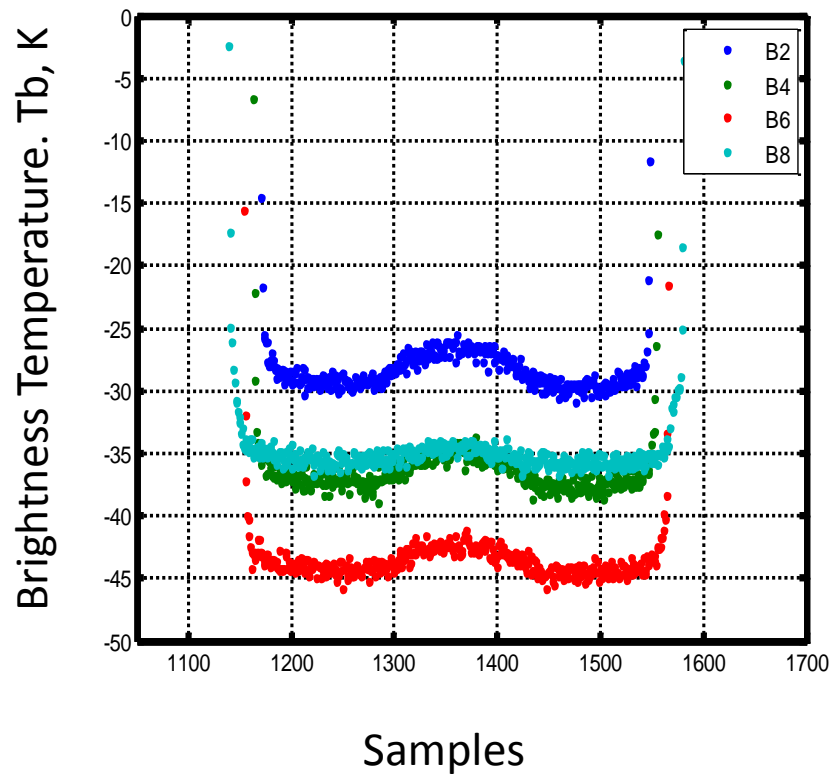


After Correction

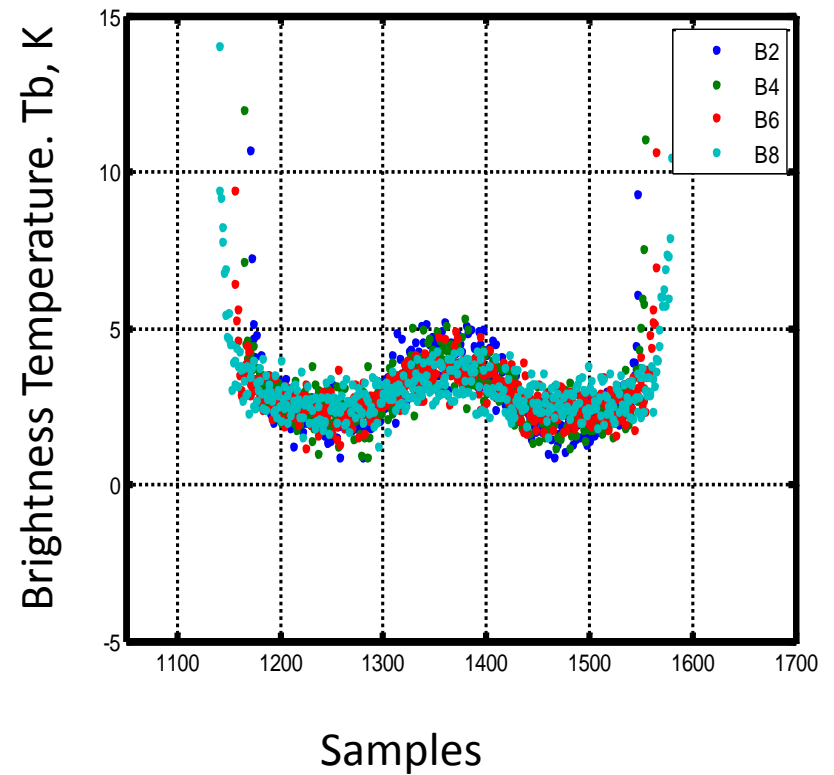


37V, Cold Sky Calibration Measurements Even Beams

Version 5.0S

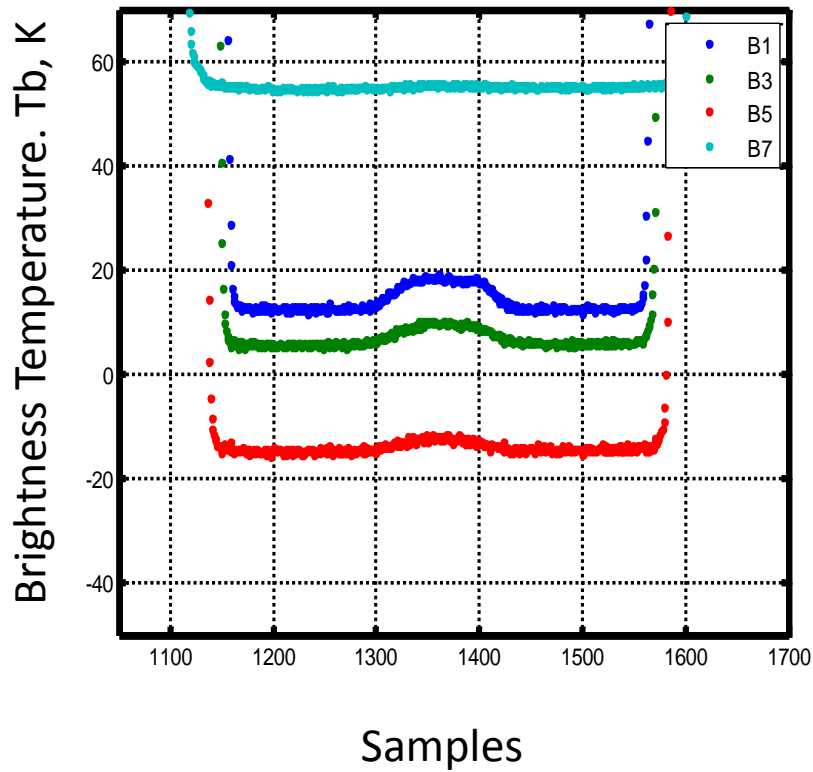


Version 6.0

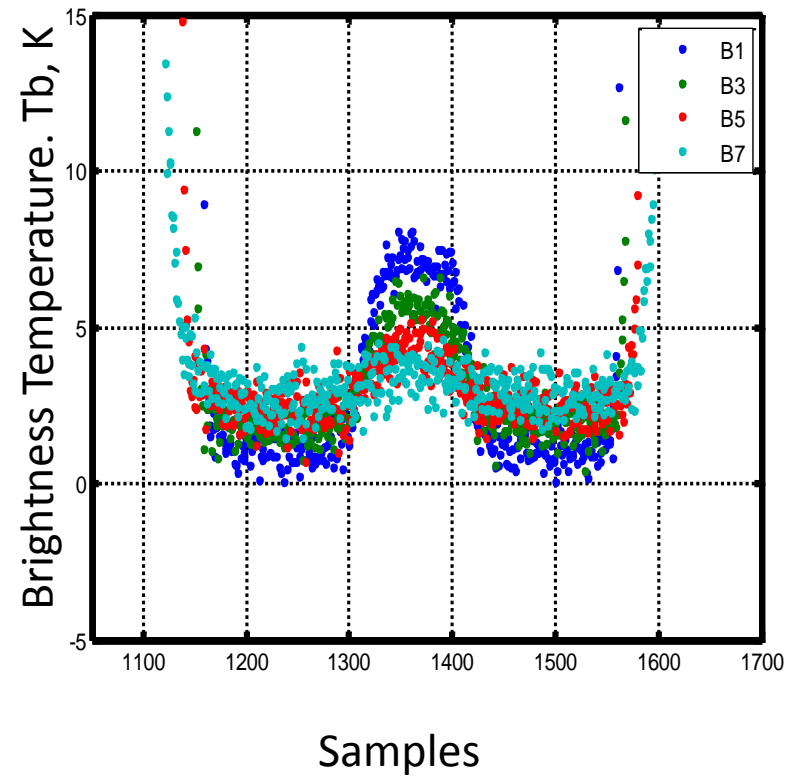


37V, Cold Sky Calibration Measurements Odd Beams

Version 5.0S

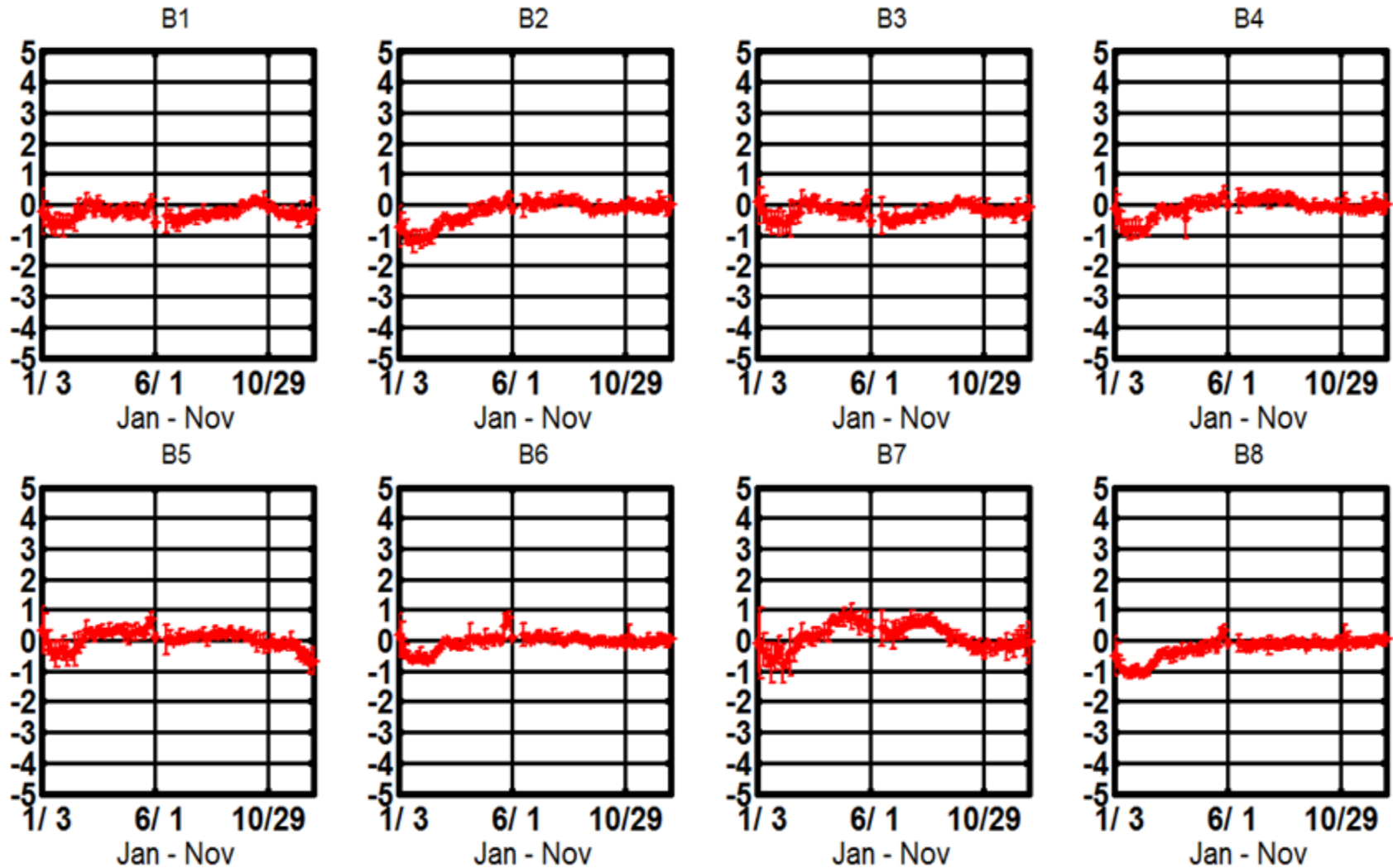


Version 6.0

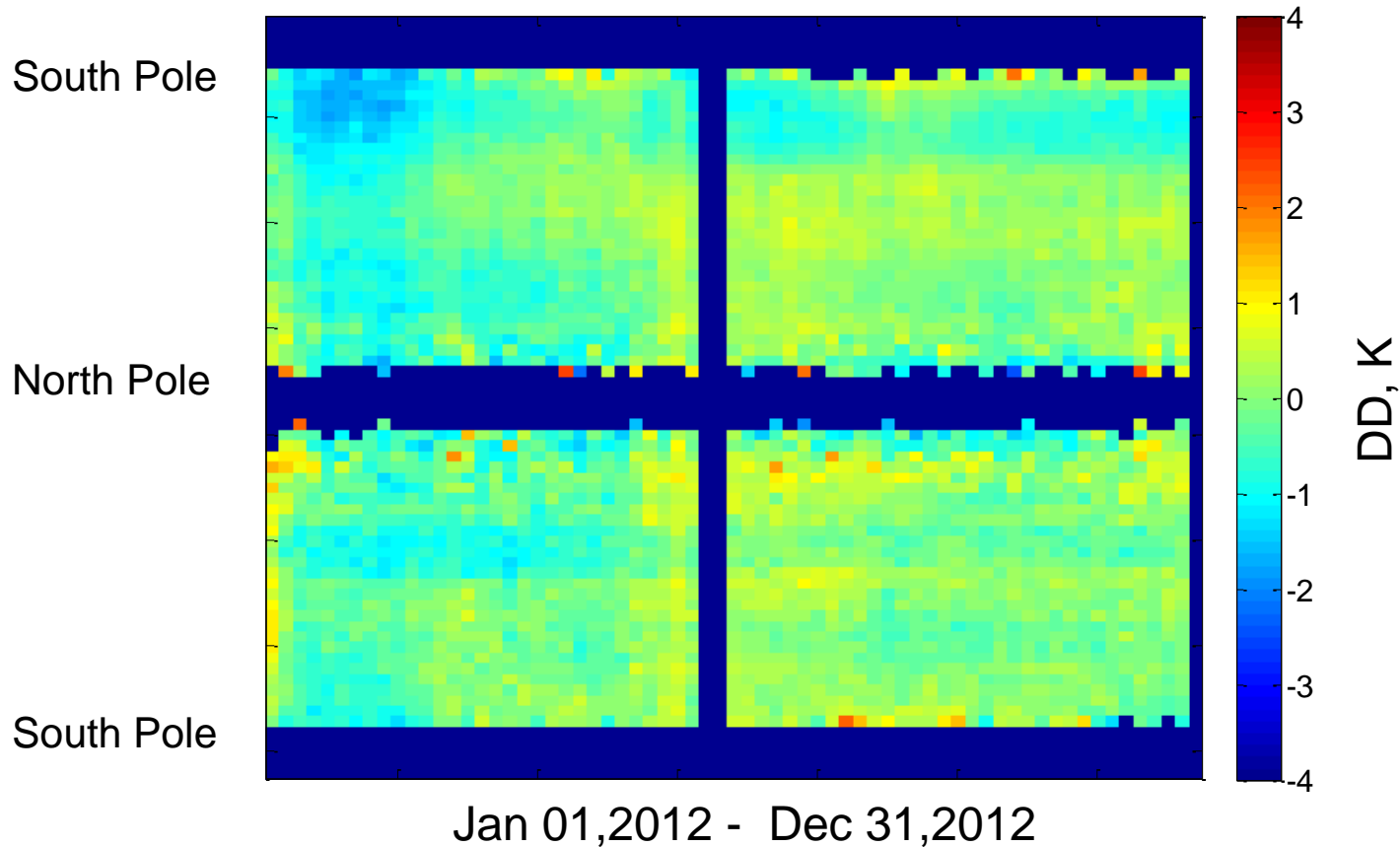


Double Difference Radiometric Biases
MWR/WindSat
(Five Days Average)
Jan 01, 2012 – Dec 31, 2012

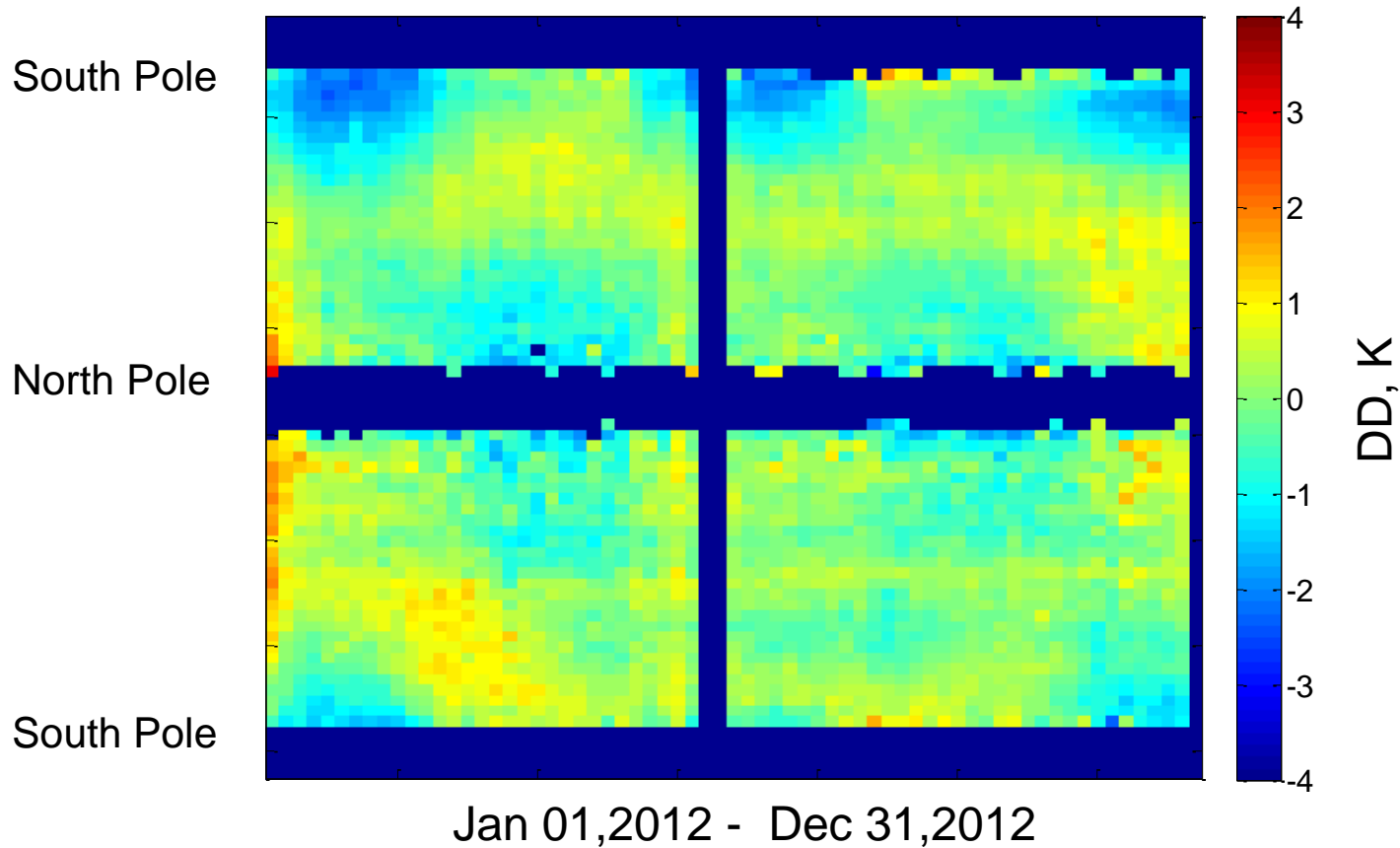
Double Difference Biases 37V (K), 2012



Five days Average in 5° Lat Zones 37V, Even Beams for 2012



Five days Average in 5° Lat Zones 37V, Odd Beams for 2012



Conclusion

- MWR Counts-to-Tb algorithm V6.0 has been developed and distributed to the AQ Cal/Val Team
 - MWR transfer function non-linearity in V5.0S has been characterized and corrected in V6.0
 - Antenna switch matrix loss coefficients were derived using re-analysis of MWR pre-launch TV calib test
- Validation of V6.0 performed using 2 years of on-orbit measurements
 - On-orbit X-CAL, between MWR and WindSat, have produced the antenna pattern correction (APC) and removed small Tb biases
- V6.0 Algorithm Theoretical Basis Document and MatLab code delivered to CONAE for science data processing

Publications

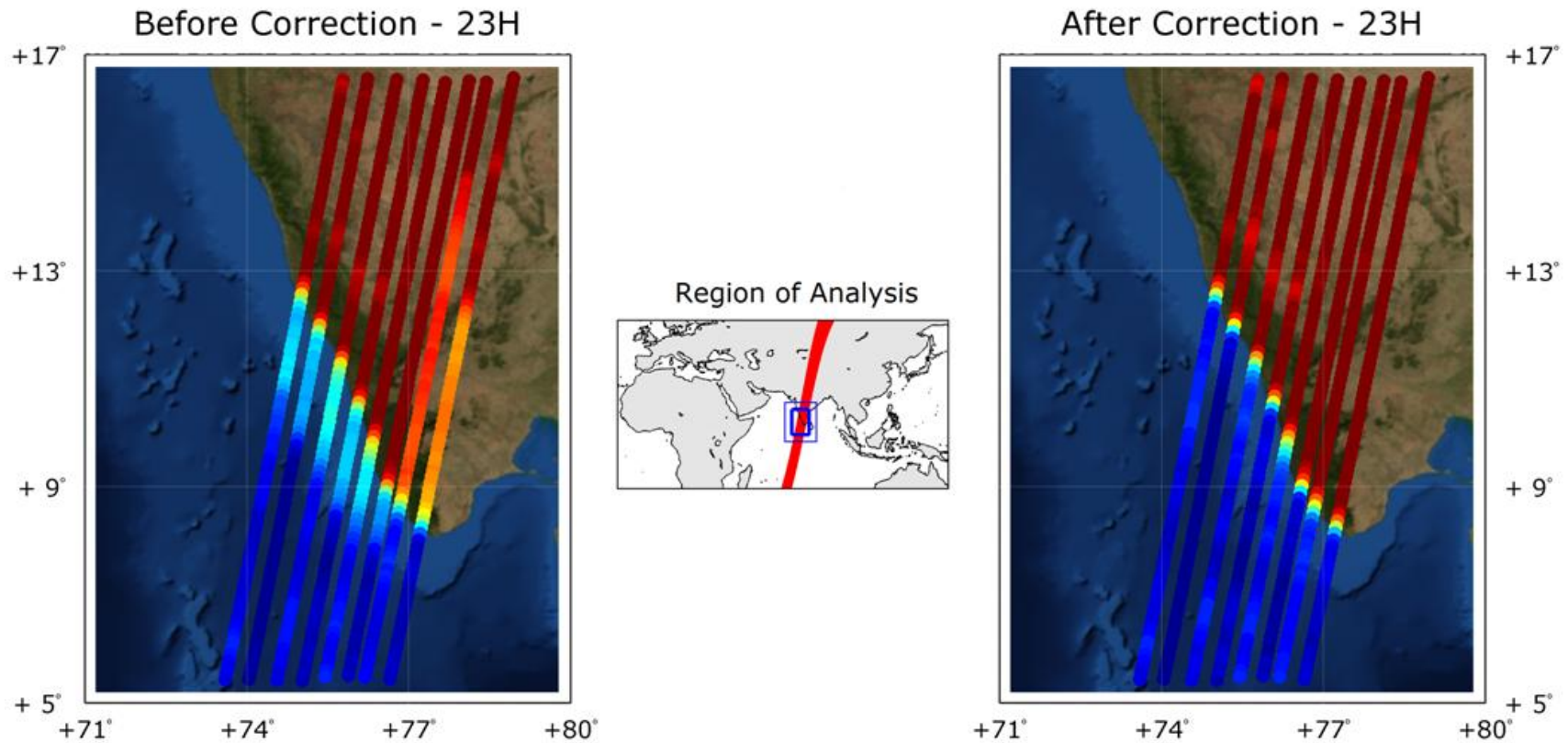
• Conferences

1. Ghazi, Z.; Biswas, S.; Jones, L.; Hejazin, Y.; Jacob, M.M., "On-orbit signal processing procedure for determining Microwave Radiometer non-linearity," Southeastcon, 2013 Proceedings of IEEE , vol., no., pp.1,5, 4-7 April 2013 doi: 10.1109/SECON.2013.6567504
2. Ghazi, Zoubair; Santos-Garcia, Andrea; Jacob, Maria Marta; Jones, Linwood, "CONAE Microwave Radiometer (MWR) counts to Tb algorithm and on-orbit validation," Microwave Radiometry and Remote Sensing of the Environment (MicroRad), 2014 13th Specialist Meeting on , vol., no., pp.207,210, 24-27 March 2014 doi: 10.1109/MicroRad.2014.6878941
3. Santos-Garcia, A; Biswas, S.; Jones, L.; Ghazi, Z.; "Aquarius/SAC-D Microwave Radiometer brightness temperature validation," Oceans, 2012 , vol., no., pp.1,4, 14-19 Oct. 2012 doi: 10.1109/OCEANS.2012.6404830

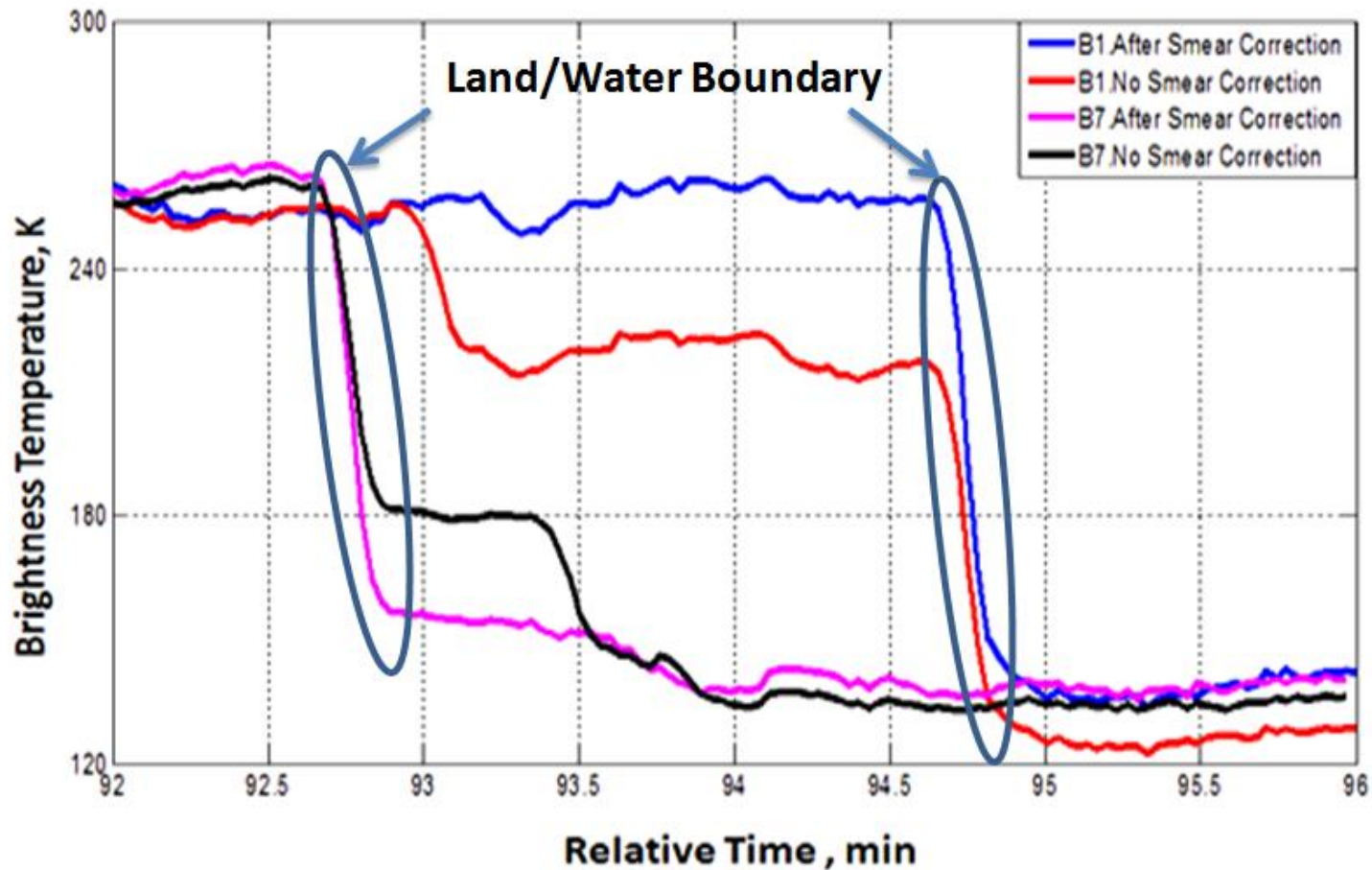
A detailed illustration of a satellite in space. The satellite is covered in gold thermal blankets and has several large black cylindrical instruments. It is positioned against the backdrop of Earth, showing blue oceans, green landmasses, and white clouds. The text "Back UP" is overlaid in a large, yellow, serif font.

Back UP

MWR Tb measurements for 8 beams of the 23 GHz H-pol channel during a descending orbital pass over the tip of India



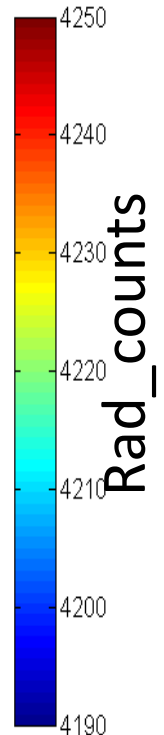
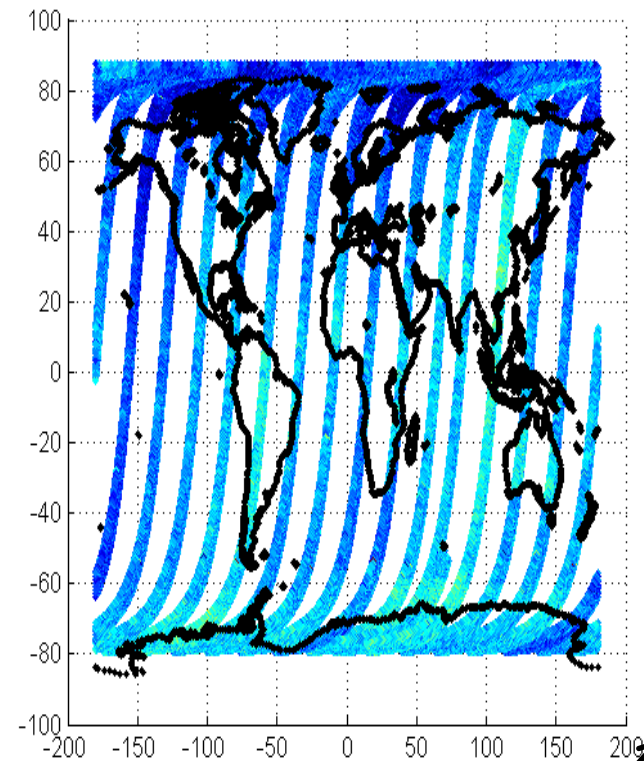
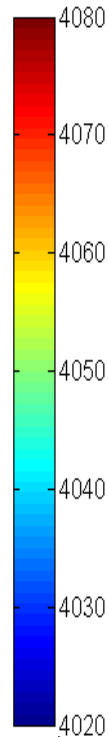
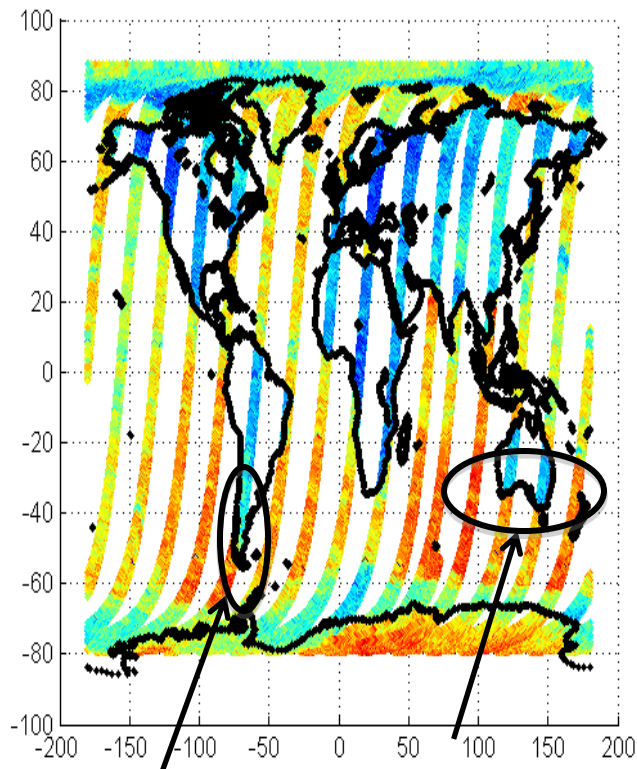
Corresponding MWR Tb time series for beam # 1 and beam # 7 of the MWR 23.8 GHz channel



37GHz V-pol, On-orbit Noise Diode Deflection (NOT gain normalized), Descending Passes for One Day (All Beams)

V5.0S

V6.0

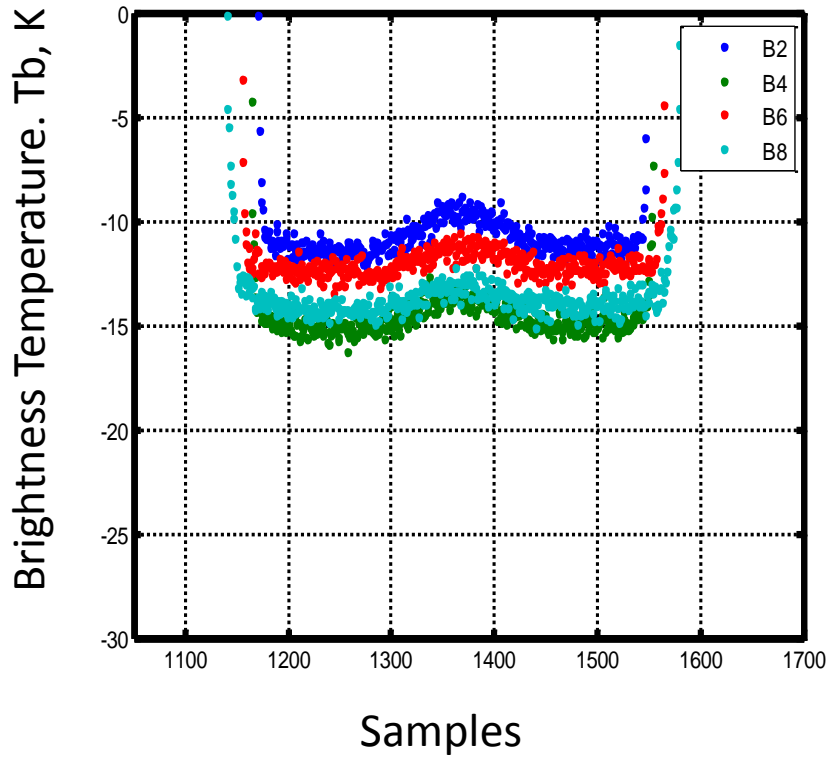


Note: "gain jumps" @
land/water boundaries

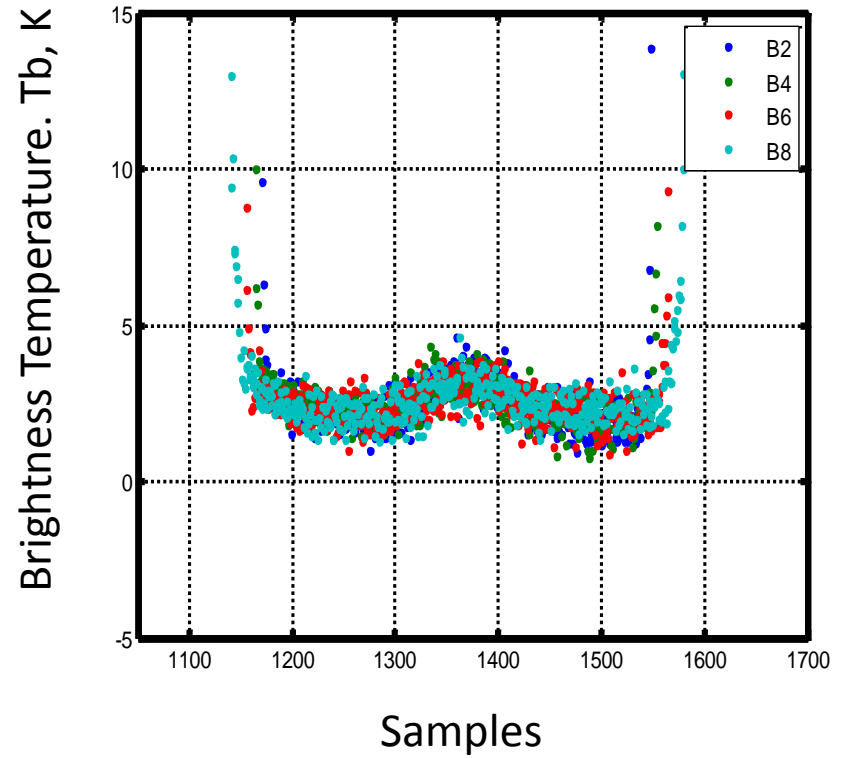
Dynamic range = 60 counts

37H Even Beams

Version 5.0S

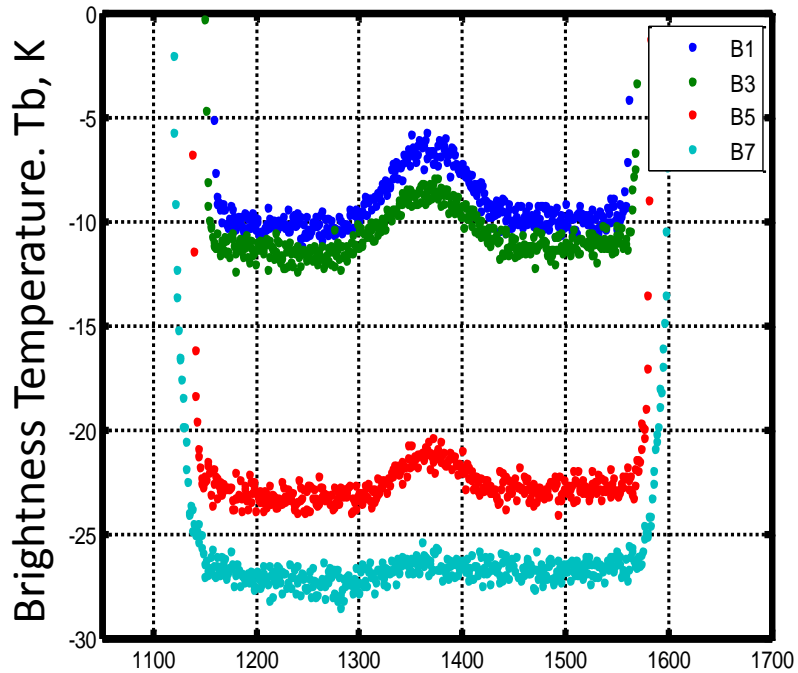


Version 6.0

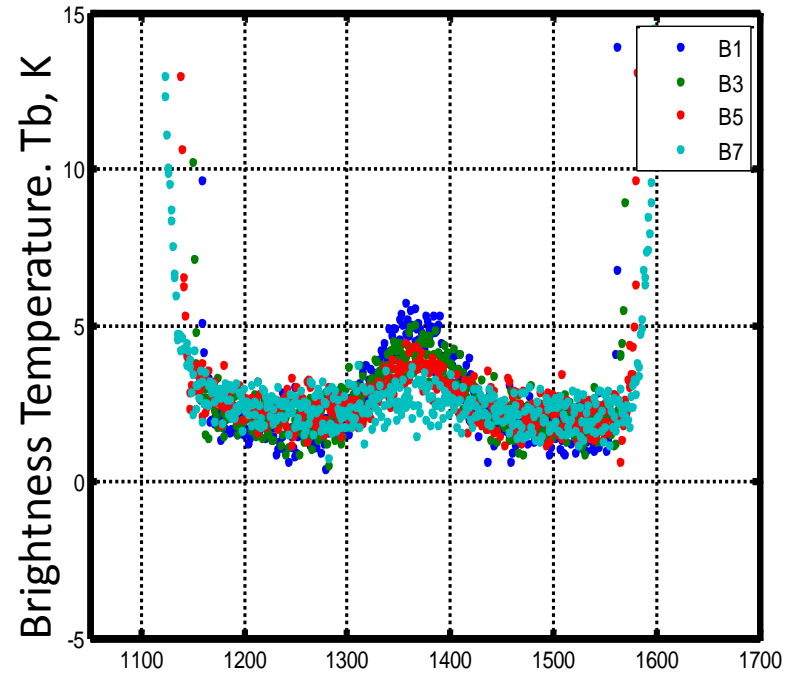


37H odd Beams

Version 5.0S

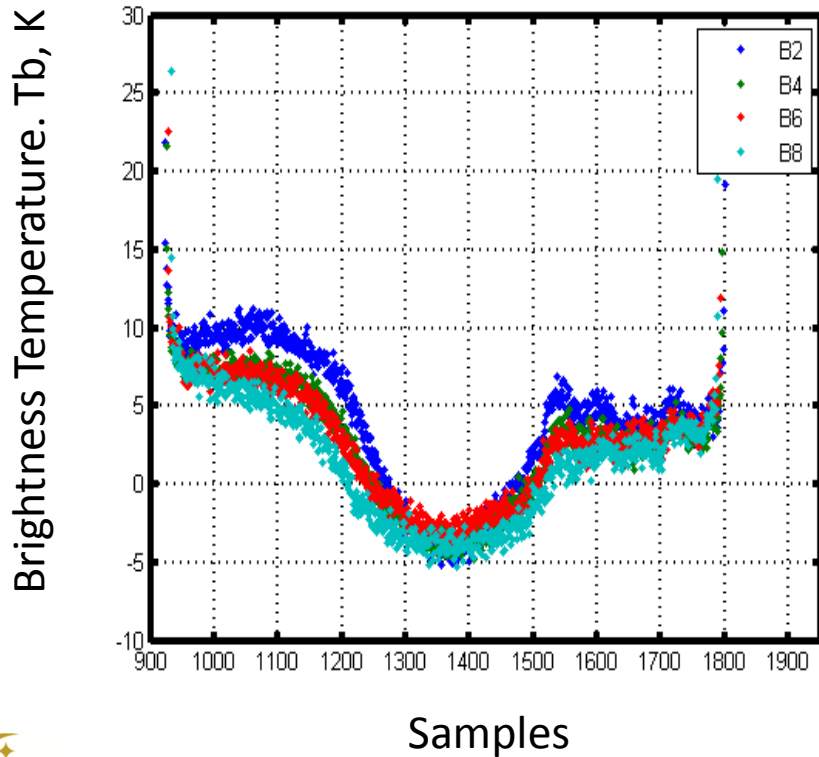


Version 6.0

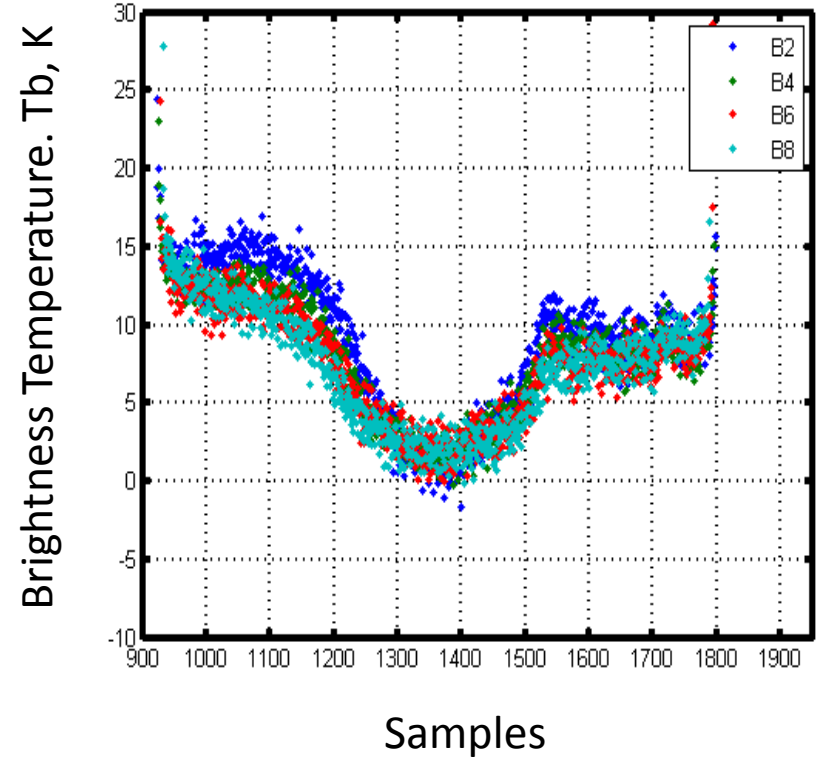


23H Even Beams

Version 5.0S

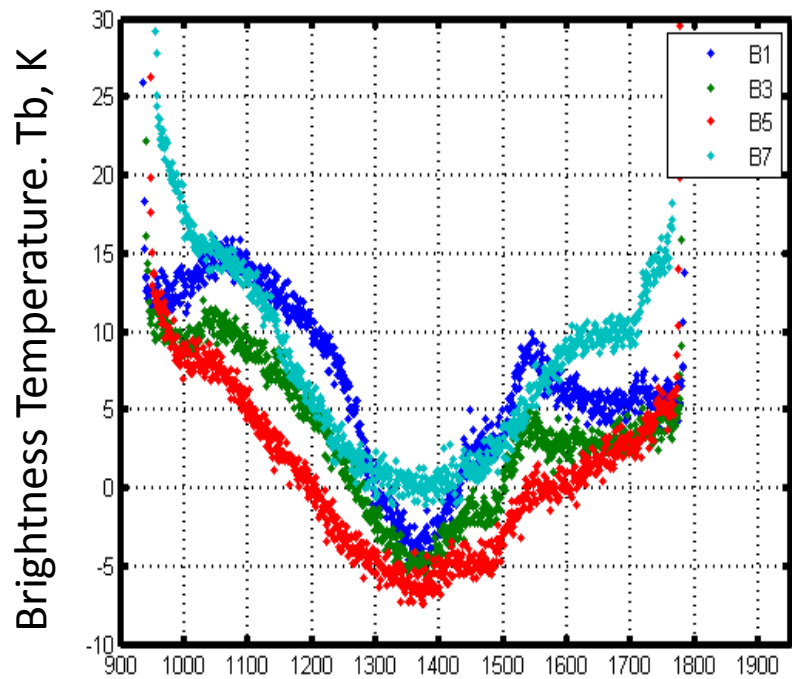


Version 6.0



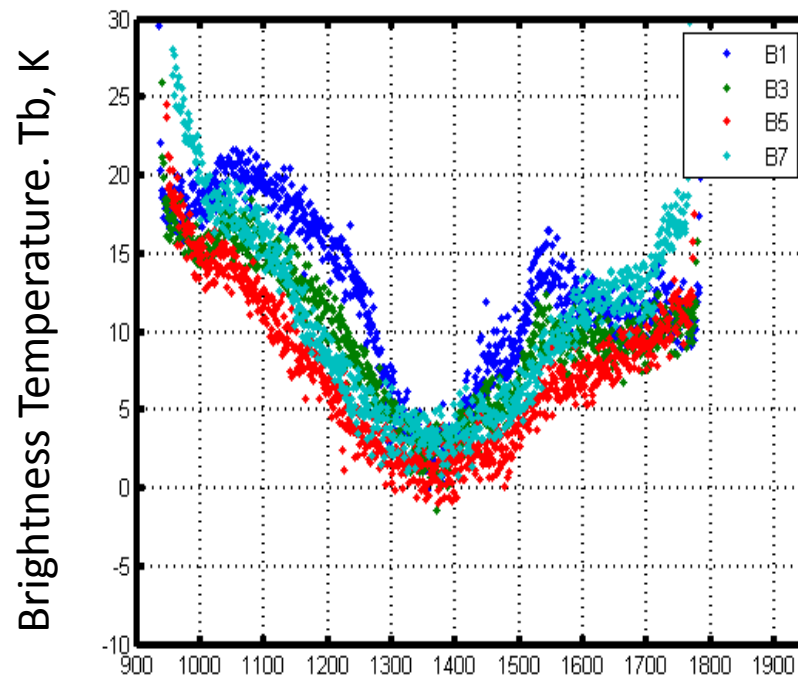
23H odd Beams

Version 5.0S



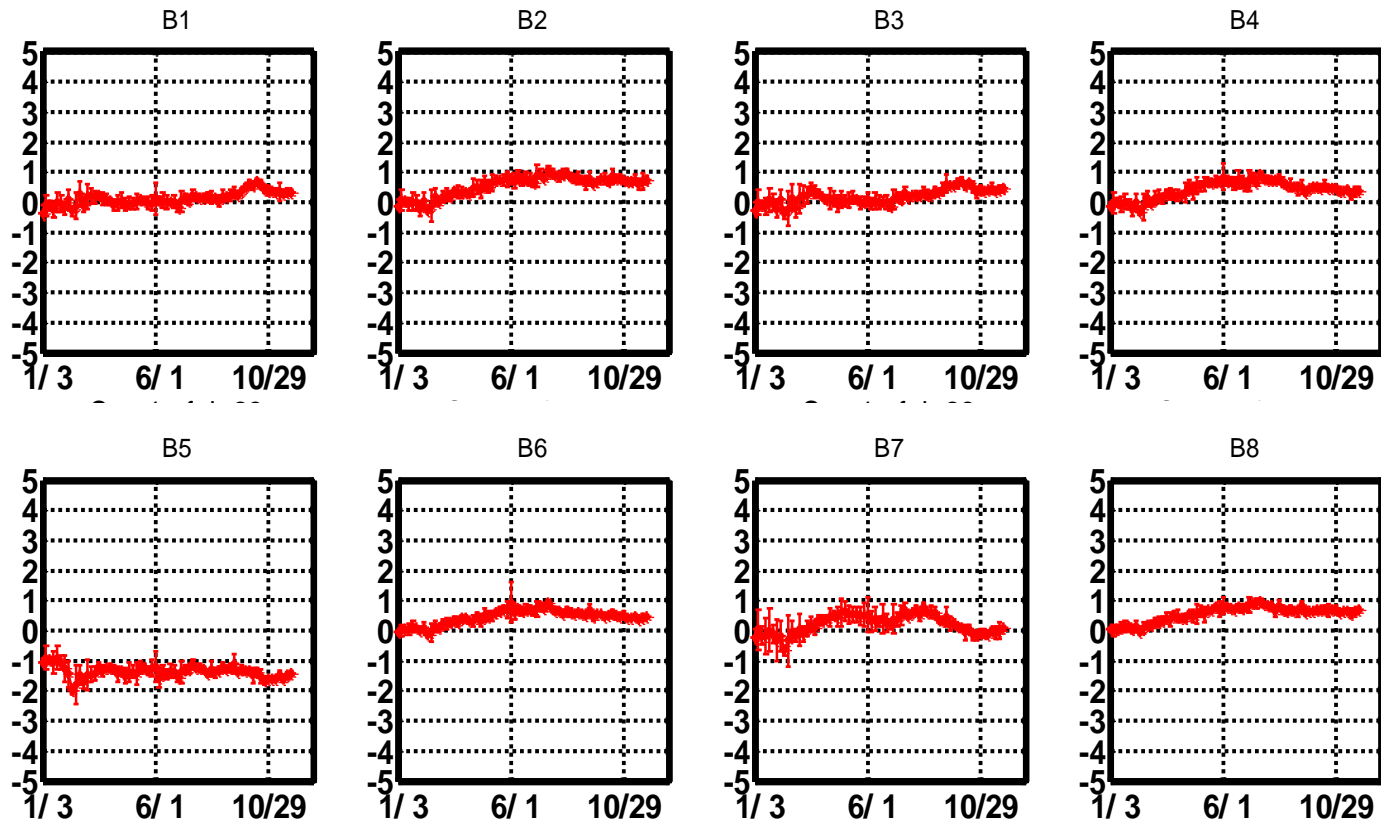
Samples

Version 6.0

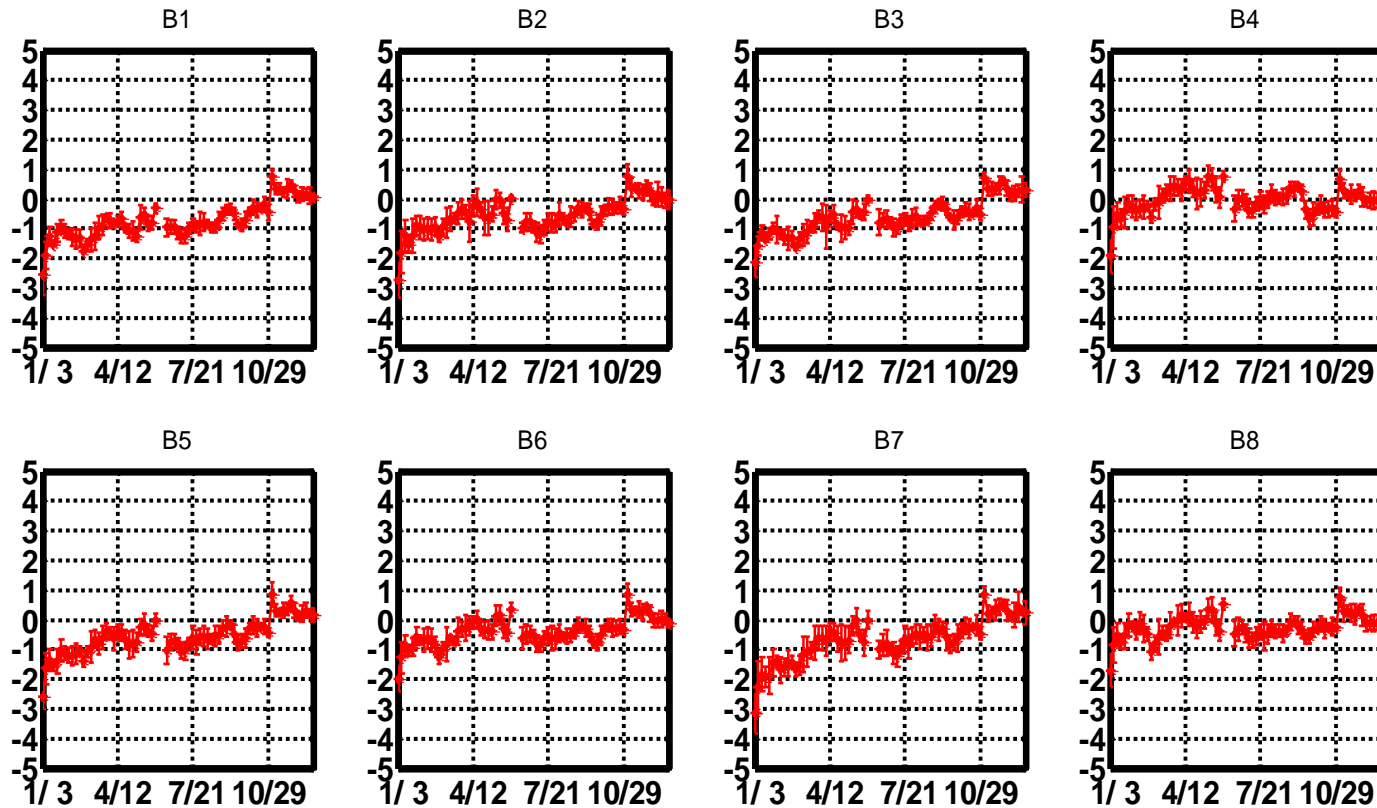


Samples

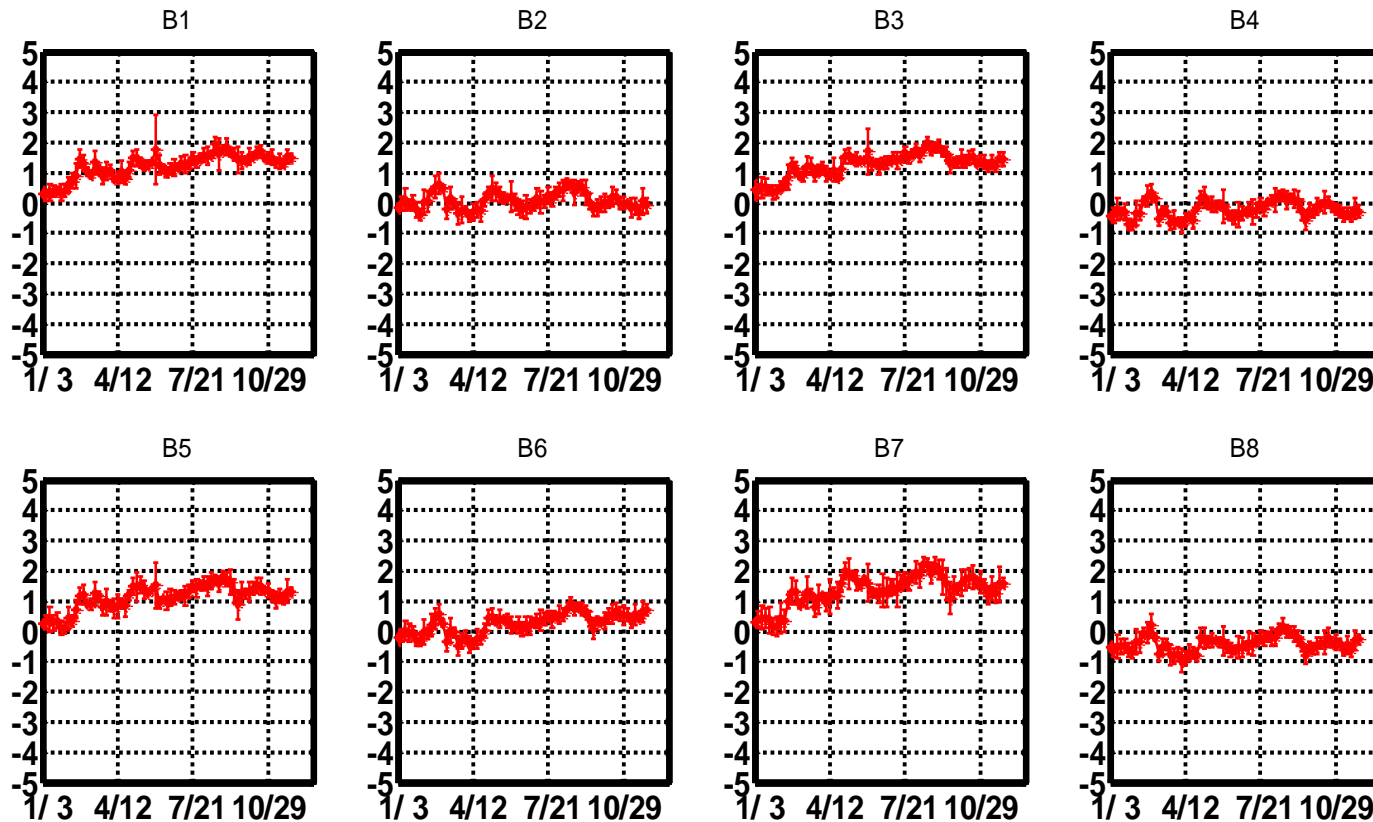
Double Difference Biases 37V (K), 2013



Double Difference Biases 37H (K), 2012

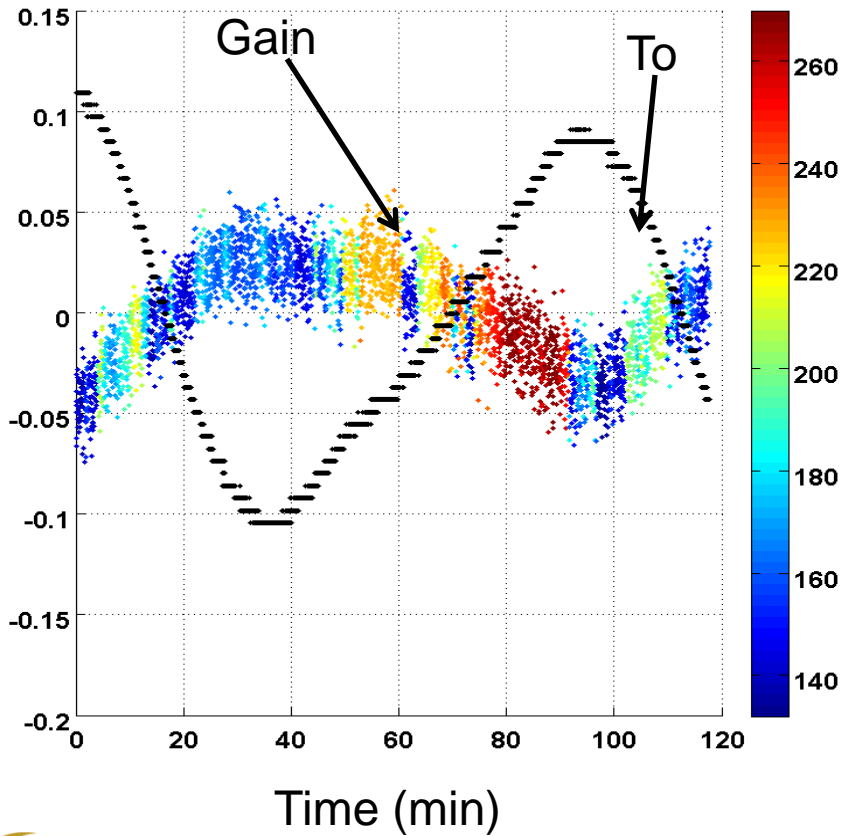


Double Difference Biases 37H (K), 2013

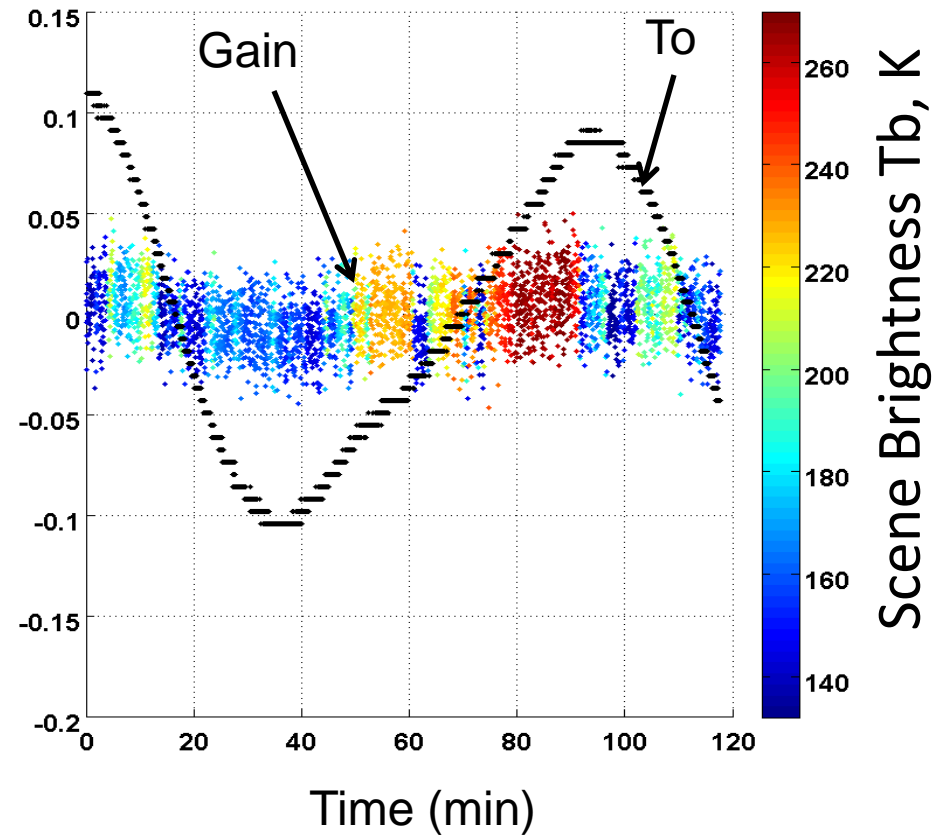


Example Gain Normalization

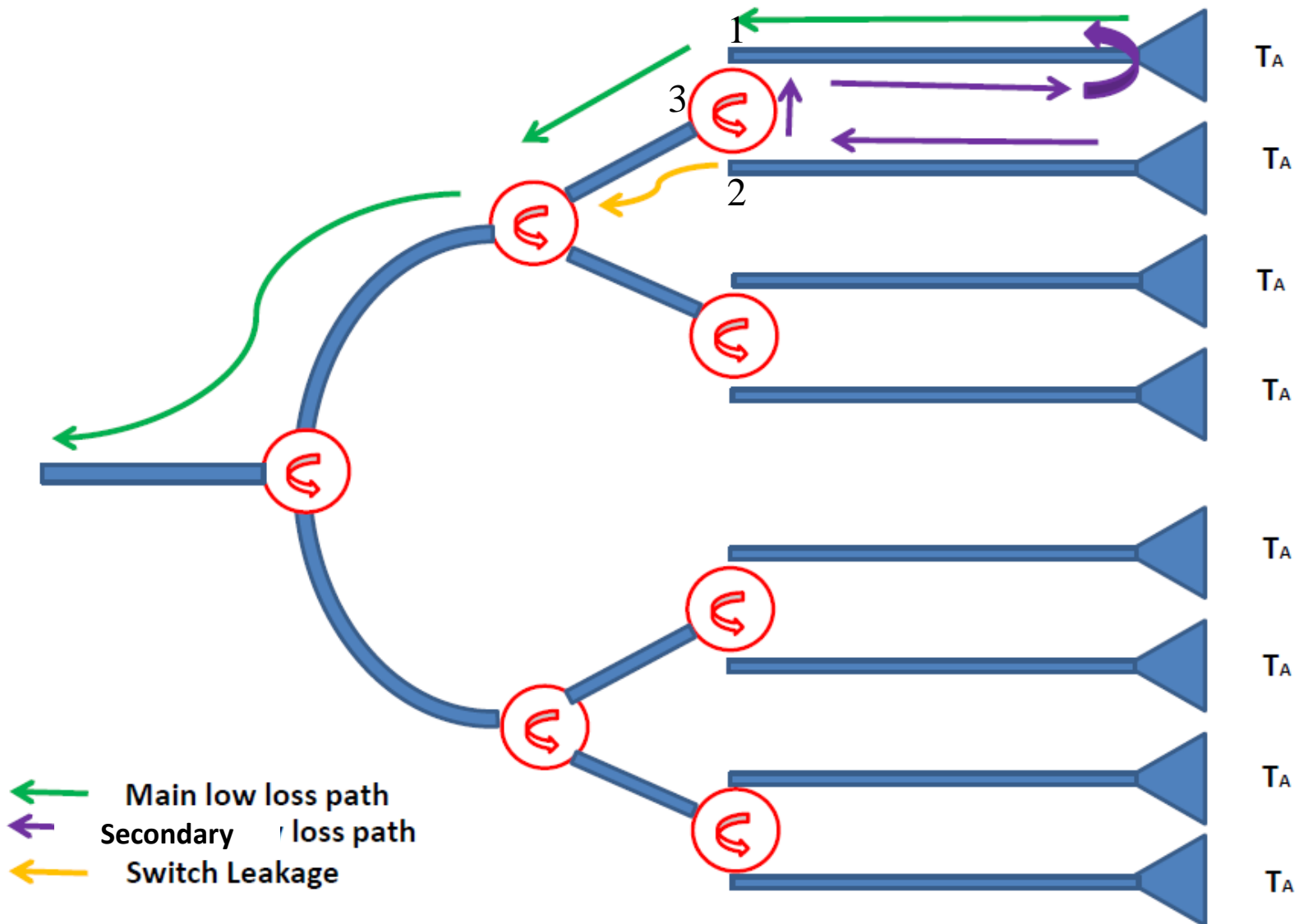
Before Gain Normalization



After Gain Normalization



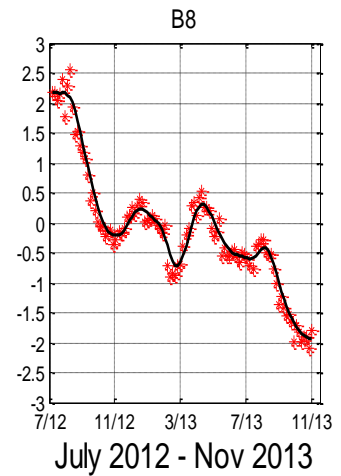
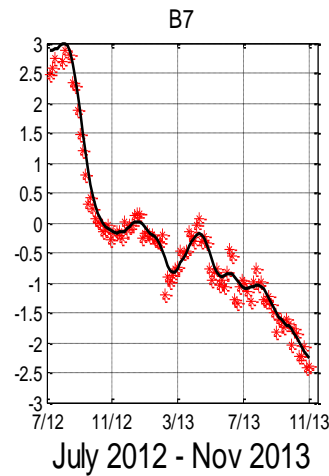
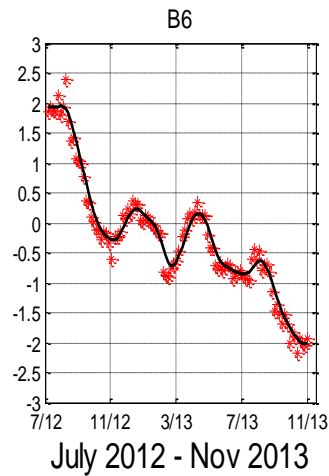
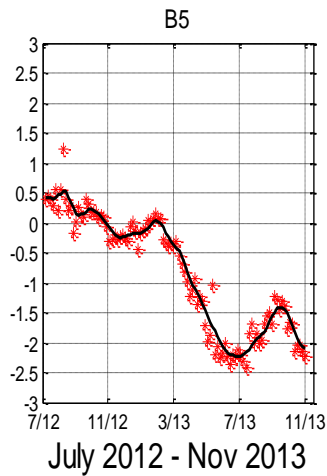
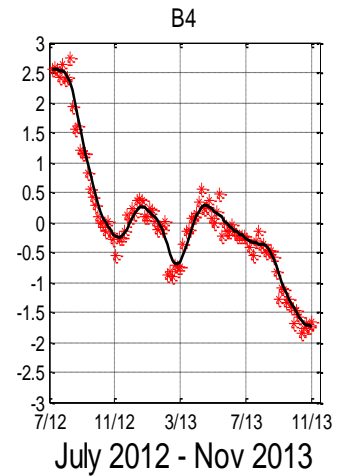
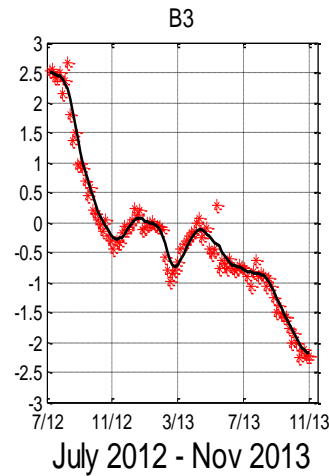
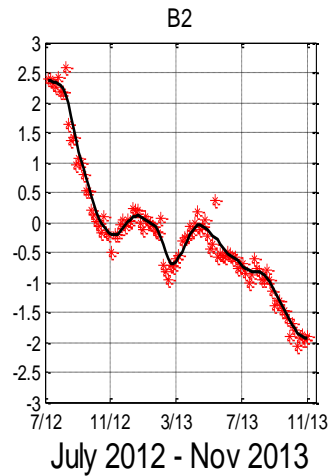
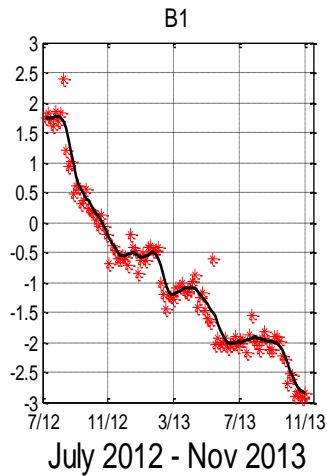
SW Matrix : Primary & Secondary Path



MWR L1B Data (V7)

- New MWR Tb data set to be used for tuning and validation of the wind speed algorithms
- XCAL 5 day double difference (DD) biases calculated between WindSat & MWR
 - **DD = MWR-WS**
- Applied triangular moving average on the 5 day DD time series to smooth the correction
- The new MWR Tb' s V7.0 = V6.0 – Tb_{biases}
 - These V7.0 Tb' s will be normalized to match the WindSat Tb' s in the mean i.e., have zero DD Tb-bias
- The new “adjusted DD” given in the following charts was derived as:
 - **DD_{adj} = DD_{V6.0} - Tb_{biases}**

V6.0 23H, DD biases (MWR-WS)



V7.0 23H DD Adjusted

